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REQUEST FOR FILING APPLICATION

Under Rule 53(a), (b) & (f)

(No Filing Fee or Oath/Declaration)

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Use for Design or Utility Applications

PATENT

APPLICATION

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Assistant Commissioner of Patents

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Washington, DC 20231

Date: Monday, April 10, 2000

Sir:

1. This is a Request for filing a new Patent Application (Design Utility) entitled:

2. (Complete) Title: A CERAMIC SUPPORT CAPABLE OF SUPPORTING A CATALYST, A CATALYST-CERAMIC BODY AND PROCESSES FOR PRODUCING SAME without a filing fee or Oath/Declaration but for which is enclosed the following:

3. Abstract 1 page(s).4. 80 Pages of Specification (only spec. and claims); 5. Specification in non-English language

6. 103 Numbered claim(s); and

7. Drawings: 1 sheet(s) 1 set informal; 8. formal of size: A4 11"9. DOMESTIC/INTERNATIONAL priority is claimed under 35 USC 119(e)/120/365(c) based on the following provisional, nonprovisional and/or PCT international application(s):

Application No.	Filing Date	Application No.	Filing Date
(1)		(2)	
(3)		(4)	
(5)		(6)	

10. FOREIGN priority is claimed under 35 USC 119(a)-(d)/365(b) based on filing in JAPAN

Application No.	Filing Date	Application No.	Filing Date
(1) 11-102547	April 9, 1999	(2) 2000-47467	February 24, 2000
(3)		(4)	
(5)		(6)	

11. (No.) Certified copy (copies): attached; previously filed (date) _____ in U.S. Application No. / filed on _____12. This is a reissue of Patent No. _____13. See top first page re prior Provisional, National, International application(s) (X box only if info is there and do not complete corresponding item 14 or 15.)14. Amend the specification by inserting before the first line -- This is a Continuation-in-Part Divisional Continuation Substitute Application (MPEP 201.09) of:14(a) National Appln. No. / filed .-- (M#)14(b) International Appln. No. PCT/ filed _____ which designated the U.S. --15. Amend the specification by inserting before the first line: --This application claims the benefit of U.S. Provisional Application No. 60/____, filed ____.--16. Extension to date: concurrently filed not needed previously filed17. Prior application is assigned to

by Assignment recorded _____ Reel _____ Frame _____

18. Attached:

19. This application is made by the following named inventor(s) :

(Double check instructions for accuracy.):

(1) Inventor	Kazuhiko	KOIKE	
First	Middle Initial	Family Name	
Residence	Japan	Japan	
City	State/Foreign Country	Country of Citizenship	
Post Office Address	c/o NIPPON SOKEN, INC., 14, Iwaya, Shimohasumi-cho, Nishio-shi, Aichi, Japan		
(include Zip Code)			

(2) Inventor	Tomohiko	NAKANISHI	
First	Middle Initial	Family Name	
Residence	Japan	Japan	
City	State/Foreign Country	Country of Citizenship	
Post Office Address	c/o NIPPON SOKEN, INC., 14, Iwaya, Shimohasumi-cho, Nishio-shi, Aichi, Japan		
(include Zip Code)			

(3) Inventor	Takeshi	UEDA	
First	Middle Initial	Family Name	
Residence	Japan	Japan	
City	State/Foreign Country	Country of Citizenship	
Post Office Address	c/o DENSO CORPORATION, 1-1 SHOWA-KARIYA-CITY AICHI-PREF. 448- 8661 JAPAN		
(include Zip Code)			

(4) Inventor	Masakazu	TANAKA	
First	Middle Initial	Family Name	
Residence	Japan	Japan	
City	State/Foreign Country	Country of Citizenship	
Post Office Address	c/o DENSO CORPORATION, 1-1 SHOWA-KARIYA-CITY AICHI-PREF. 448- 8661 JAPAN		
(include Zip Code)			

(5) Inventor			
First	Middle Initial	Family Name	
Residence			
City	State/Foreign Country	Country of Citizenship	
Post Office Address			
(include Zip Code)			

20. NOTE: FOR ADDITIONAL INVENTORS, check box and attach sheet with same information regarding additional inventors.

Pillsbury Madison & Sutro LLP
Intellectual Property Group

By: Atty. Timothy J Klima

Reg. No. 34852

Sig: _____

Fax: (202) 822-0944
Tel: (202) 861-3662

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1100 New York Avenue, NW.
Ninth Floor
Washington, DC 20005-3918
Tel: (202) 861-3000
Atty/Sec: TJK/nlh

A CERAMIC SUPPORT CAPABLE OF SUPPORTING A CATALYST,
A CATALYST-CERAMIC BODY AND PROCESSES FOR PRODUCING SAME

5 BACKGROUND OF THE INVENTION

1. Field of the Invention

The present invention relates to a ceramic support for supporting a catalyst component for cleaning an exhaust gas from an internal combustion engine of an automobile, etc., a catalyst-ceramic body using the ceramic support, and processes for producing these

10 2. Description of the Related Art

A widely used conventional catalyst support for a catalyst component used to clean an exhaust gas is a honeycomb structure of cordierite, which is excellent in thermal shock resistance, the surface of which is coated with γ -alumina. This is because a cordierite honeycomb structure has a relatively small specific surface area such that a sufficient amount of a catalyst component cannot be supported thereon. Therefore, γ -alumina having a large specific surface area is coated on the surface of a cordierite honeycomb structure, to support a catalyst component such as Pt thereon.

Because recent changes in the exhaust gas regulation, a quick activation of a catalyst is required to reduce hydrocarbons from an exhaust gas generated immediately after the starting of an engine. To satisfy to this requirement, a reduction of the thermal capacity of a catalyst support, in order to increase the temperature elevating rate of a catalyst to an activated state has been considered. Reduction of the thermal capacity of a catalyst support can be made by decreasing the thickness of cell walls of a cordierite honeycomb structure. However, even if the thickness of cell walls of a cordierite honeycomb structure is reduced, to have a small thermal capacity, the amount of a coating of γ -

alumina does not decrease, resulting in lowering the effect of reducing the thermal capacity of the honeycomb structure as a whole.

In reducing the thickness of cell walls of a cordierite honeycomb structure, a reduction in pressure loss can be expected but this effect is also reduced by a coating of γ -alumina. Further, although a cordierite honeycomb structure itself has a small thermal expansion coefficient, about $0.5 \times 10^{-6}/^{\circ}\text{C}$, the thermal expansion coefficient of a honeycomb structure with a coating of γ -alumina increases to about $1.5 \times 10^{-6}/^{\circ}\text{C}$, by which the thermal shock resistance of the honeycomb structure significantly decreases.

In order to solve the above problems, an increase in the specific surface area of a cordierite honeycomb structure to eliminate the necessity of a coating of γ -alumina has been considered. For example, Japanese Examined Patent Publication (Kokoku) No. 5-50338 discloses a method of carrying out acid and heat treatments of a cordierite honeycomb structure to increase the specific surface area thereof. Although it is mentioned that this method allows increase in the specific surface area from $1 \text{ m}^2/\text{g}$ to about $30 \text{ m}^2/\text{g}$, this method disadvantageously decreases the strength of the cordierite honeycomb structure. Since the acid treatment selectively dissolves MgO and Al_2O_3 from the cordierite to increase the surface area, the crystal lattice of cordierite itself is destroyed and the strength of the cordierite honeycomb structure decreases.

A honeycomb structure should have a compressive strength in the direction of the flow channel as a high as 10 MPa or more to be installed in an exhaust gas cleaning catalyst converter. In the method disclosed in JPP'338, a compressive strength, in the direction of the flow channel, of 10 MPa or more cannot be attained if the treated cordierite honeycomb structure has a cell wall

thickness of less than 150 μm at a cell density of 400 cpsi (cells per square inch). However, a cordierite honeycomb structure having a cell wall thickness of 150 μm at a cell density of 400 cpsi has a thermal capacity equivalent to that of a cordierite honeycomb structure having a cell wall thickness of 120 μm at a cell density of 400 cpsi and coated with γ -alumina at 180 g/l. Therefore, the above method in JPP'338 does not provide an improved effect in reducing the thermal capacity of a catalyst support, considering the fact that a cordierite honeycomb structure having a cell wall thickness of 85 μm at a cell density of 400 cpsi can be produced at present.

It is therefore required that a cordierite honeycomb structure for cleaning an exhaust gas has a compressive strength in the direction of the flow channel of 10 MPa or more with a cell wall thickness of 120 μm or less at a cell density of 400 cpsi and can support a required amount of a catalyst component without a coating of γ -alumina.

Thus, the object of the present invention is to make a ceramic support itself which can support a required amount of a catalyst component, by which the necessity of coating with γ -alumina is eliminated, and to thereby provide a ceramic support to be used to support a catalyst component for cleaning an exhaust gas from an internal combustion engine of an automobile, etc., which has a reduced thermal capacity and pressure loss as well as a lower thermal expansion coefficient. Also, the object of the present invention is to provide a catalyst-ceramic body and processes for producing the ceramic support and the catalyst-ceramic body.

SUMMARY OF THE INVENTION

The present inventors considered the conditions for fine pores necessary to directly support a catalyst

component on a ceramic support and found that fine pores formed by defects such as oxygen vacancies and lattice defects in the ceramic crystals, by fine cracks formed on the surface of a ceramic support or body, and by
5 vacancies of elements constituting the ceramic, etc. can make a ceramic support to support a required amount of a catalyst component without a coating of γ -alumina. It was also found that it is desired that the number of the above fine pores is $1 \times 10^{11}/\text{l}$ or more, more preferably
10 $1 \times 10^{16}/\text{l}$, further preferably $1 \times 10^{17}/\text{l}$ or more. These fine pores may be present alone or in combination.

The ceramic support of the present invention is preferably a cordierite honeycomb structure comprising cordierite, the theoretical composition of which is
15 $2\text{MgO} \cdot 2\text{Al}_2\text{O}_3 \cdot 5\text{SiO}_2$, as the main component and having a honeycomb shape. Since the diameter of an ion of a catalyst component is usually in the order of about 0.1 nm, a cordierite honeycomb structure preferably has a diameter or width of the fine pores formed on the surface
20 which is up to 1000 times, preferably 1 to 1000 times the above ion diameter or about 0.1 nm, that is, 0.1 to 100 nm, and a depth of the fine pores which is not smaller than a half of the diameter of an ion of a catalyst component, that is, 0.05 nm or more. By having
25 the above predetermined number of such fine pores, the ceramic support of the present invention can directly support a required amount of a catalyst component with a required strength being maintained.

When the ceramic support has the fine pores of
30 oxygen vacancies or lattice defects, the number of the fine pores significantly depends on the amount of the oxygen in the cordierite honeycomb structure. To have the above stated number of the fine pores, it is suggested that the content of the oxygen in the
35 cordierite honeycomb structure is made less than 47% by weight or more than 48% by weight. Also, it is suggested

that the lattice constant of the b_0 axis of the cordierite crystal is made larger than 16.99 or smaller than 16.99. Specifically, the cordierite honeycomb structure preferably includes cordierite crystals having 5 one or more in number of at least one of the oxygen vacancy and lattice defects in the unit lattice of cordierite crystal (also called as the unit cell of cordierite crystal), in an amount of not less than $4 \times 10^{-8}\%$ of said ceramic, more preferably not less than 10 $4 \times 10^{-5}\%$ of said ceramic; or preferably has cordierite crystals which comprise not less than 4×10^{-8} , more preferably not less than 4×10^{-7} , of at least one of oxygen vacancy and lattice defect per the unit lattice of cordierite crystal, to make the number of the fine pores 15 of the ceramic support $1 \times 10^{16}/\text{l}$ or more, preferably $1 \times 10^{17}/\text{l}$ or more.

A catalyst component is usually supported on a ceramic support by immersing a ceramic support in a solution of catalyst component ions in a solvent. A 20 conventional cordierite honeycomb structure coated with γ -alumina has a typical size of fine pores of γ -alumina of about 2 nm, while the catalyst metal particles typically have a particle size of about 5 nm, larger than the size of the fine pores of γ -alumina. Therefore, it 25 is considered that the fine pores of γ -alumina are necessary to support catalyst component ions, rather than to support metal particles. Thus, if a ceramic support has fine pores with a diameter or width equivalent to or larger than the diameter of catalyst component ions, that 30 is, fine pores with a diameter or width of 0.1 nm or more, it is possible to support catalyst component ions. However, the size of the fine pore of the ceramic support should be not larger than 1000 times the diameter of catalyst component ions, in order to have a sufficient

strength of the honeycomb structure. This size is 100 nm or smaller when the diameter of catalyst component ions is 0.1 nm. It is sufficient for supporting catalyst component ions that the depth of the fine pores be a half or more than the diameter of catalyst component ions.

Since the fine pores of defects or cracks are very fine and the surface area of a ceramic support having them cannot be measured by the usual measuring methods, the present invention designates or states the number of fine pores necessary to support a predetermined amount of a catalyst component. The amount of a catalyst metal supported in a presently used three way catalyst is about 1.5g per liter of the volume of the honeycomb structure. The diameter of the catalyst metal particles should be about 1000 nm or smaller, preferably about 20 nm or smaller to exhibit the exhaust gas cleaning capability. If it is assumed that the weight of a catalyst metal supported by a honeycomb structure with a volume of 1 liter is W grams and all the catalyst metal particles are spheres with a diameter of r cm, the number N of the catalyst metal particles supported can be expressed by the following formula (1), in which ρ stands for the density of the catalyst metal in g/cm^3 :

$$N \times (4/3)\pi r^3 \times \rho = W \quad (1)$$

In the case that Pt, which is the catalyst metal used in a presently used three way catalyst-ceramic body for cleaning, is supported at 1.5 g/l and all the Pt particles have a diameter of 1000 nm, the number of the Pt particles supported is $1.34 \times 10^{11}/\text{l}$ from the above formula (1), in which the density of Pt is $21.45 \text{ g}/\text{cm}^3$. Since each catalyst metal particle requires about one fine pore for supporting, the number of the fine pores necessary to support a required amount of the catalyst metal particles directly on a ceramic support is at least $1 \times 10^{11}/\text{l}$ or more. When the average diameter of the catalyst metal particles becomes about 10 nm, the

cleaning capability can be about equal to that of a presently used three way catalyst. At that time, the number of the catalyst metal particles is $1.34 \times 10^{17}/l$ from the above formula (1) and, therefore, the number of 5 necessary fine pores is preferably $1 \times 10^{17}/l$ or more.

On the other hand, the weight of a cordierite honeycomb structure having a cell wall thickness of 100 μm at a cell density of 400 cpsi is about 230 g/l. If all the honeycomb structure is made of cordierite 10 $Mg_2Al_4Si_5O_{18}$, the number of the unit lattices of cordierite crystal (i.e., the number of the unit cells of cordierite crystal, equal to the number of cordierite molecules) can be calculated by the following formula (2):

$$\begin{aligned} & (\text{number of unit lattices of cordierite}) \\ 15 & = (\text{weight of honeycomb}) / (\text{molecular weight of} \\ & \quad \text{cordierite}) \times (\text{Avogadro number}) \\ & = ((230 \text{ g/l}) / 584.95\text{g}) \times 6 \times 10^{23} \\ & = 2.36 \times 10^{23}/l \end{aligned} \tag{2}$$

If it is assumed that a cordierite honeycomb 20 structure has $1 \times 10^{16}/l$ of oxygen vacancies or lattice defects and only one such defect of oxygen vacancy or lattice defect is formed in one cordierite crystal unit lattice, the percent ratio of cordierite unit lattices having a defect in relation to the cordierite crystal as 25 a whole can be calculated by the following formula (3):

$$\begin{aligned} & (\text{percent ratio of cordierite unit lattices having a} \\ & \quad \text{defect}) \\ 30 & = (\text{number of defects}) / (\text{number of unit lattices of} \\ & \quad \text{cordierite}) \\ & = (1 \times 10^{16} / 2.36 \times 10^{23}) \times 100\% \\ & = 4 \times 10^{-8} \times 100\% \\ & = 4 \times 10^{-6}\% \end{aligned}$$

Also, if the number of defects is 1×10^{17} , the percent ratio of cordierite unit lattices of having a

defect to all the cordierite crystals becomes $4 \times 10^{-5}\%$.

The number of defects such as oxygen vacancies or lattice defects can be also calculated by the formula: (number of defects)/(number of cordierite crystals),
5 similar to the formula (3). That is, if the number of defects is $1 \times 10^{16}/1$, the number of defect per unit crystal lattice is 4×10^{-8} . If the number of defects is $1 \times 10^{17}/1$, the number of defects per unit crystal lattice is 4×10^{-7} .

10 In accordance with the present invention, the catalyst supporting capability is provided to a cordierite honeycomb structure by methods including (1) a method of forming an oxygen vacancy, or a lattice defect such as a metal vacant point and a lattice strain in a
15 cordierite crystal unit lattice, (2) a method of forming a large number of fine cracks in at least one of the amorphous and the crystal phases, (3) a method of dissolving cordierite constituting elements or impurities in a liquid phase method to form vacancies or pores, (4)
20 a method of chemically or physically forming vacancies or pores in a vapor phase method, and (5) a method of incorporating an oxygen storage substance. In the present invention, a cordierite honeycomb structure which has a predetermined number of the fine pores formed as
25 above can directly support a catalyst component. Also, since these fine pores formed by the methods as above do not cause damage to the ceramic crystal lattice as in the prior art method, the cordierite honeycomb structure can have a compressive strength in the direction of the flow
30 channel of 10 MPa or more and a thermal expansion coefficient of $1 \times 10^{-6}^{\circ}\text{C}$ or less even if the thickness of the cell wall is reduced.

35 The cordierite honeycomb structures provided with a catalyst supporting capability by the above methods are described below.

First, a cordierite honeycomb structure, in which the cordierite crystal thereof has an oxygen vacancy or a lattice defect such as a metal vacancy or lattice strain, is described. The defect which makes catalyst component supporting possible, includes an oxygen vacancy or lattice defect. The oxygen vacancy is a defect formed when oxygen is in an amount insufficient to compose a cordierite crystal lattice, and a catalyst component can be supported on a pore which is formed where the oxygen is lacked. In order to support a required amount of a catalyst component, it is suggested that the content of oxygen in a honeycomb structure is made to be less than 47% by weight.

The lattice defect is a defect formed when oxygen is incorporated in an amount excess to compose a cordierite crystal lattice, and a catalyst component can be supported on a pore which is formed by a strain of crystal lattice or a metal vacant point. In order to support a required amount of a catalyst component, it is suggested that the content of oxygen in a honeycomb structure is made to be more than 48% by weight.

A cordierite honeycomb structure having a defect in the crystal lattice can be produced by controlling the atmosphere for firing the honeycomb structure or by using particular cordierite materials. Among them, the oxygen vacancy can be formed (i) by controlling the firing atmosphere to a reduced pressure atmosphere or a reducing atmosphere, (ii) by using a compound not containing oxygen in at least part of the cordierite materials and firing the cordierite materials in a low oxygen concentration atmosphere, and (iii) by replacing some of at least one other-than-oxygen element constituting the cordierite by an element having a valency smaller than the replaced element. The lattice defect can be formed (iv) by replacing some of at least one element other than oxygen constituting the cordierite by an element having a valency larger than the replaced element. These methods

(i) to (iv) are described in the following.

In the method (i) of forming a cordierite honeycomb structure having oxygen vacancies, the cordierite materials may be those generally used for producing
5 cordierite, for example, talc ($Mg_3Si_4O_{10}(OH)_2$), kaolin ($Al_2Si_2O_5(OH)_4$), calcined kaolin, alumina (Al_2O_3), aluminum hydroxide ($Al(OH)_3$), etc. In addition to the above,
10 oxides, hydroxides and the like which contain at least one of Si, Al and Mg elements constituting cordierite may be used as an Si source, an Al source and a Mg source.

The above cordierite materials are formulated as to form a theoretical composition as described before, and combined with a shaping agent such as a binder, a lubricant and a humidity-keeping agent as well as water,
15 which is then extruded to form a honeycomb shape. The shaped body is heated in air at about 500°C to remove an organic binder, followed by firing in a reduced pressure or reducing atmosphere to form a honeycomb structure. In firing in a reduced pressure atmosphere, the pressure is
20 preferably about 4000 Pa (30 Torr) or less and the firing is typically carried out at about 1350°C or higher for 2 hours or more.

As firing in a reduced pressure atmosphere, the oxygen contained in the cordierite materials leaves as a gas during the firing so that the oxygen becomes
25 insufficient to compose cordierite and an oxygen vacancy in the cordierite crystal unit is formed. When firing in a reducing atmosphere, that is, when the firing is carried out under the conditions similar to the above but
30 in a reducing atmosphere such as a hydrogen atmosphere, the oxygen contained in the cordierite materials reacts with the reducing gas to leave as a gas during the firing. As a result, the oxygen becomes insufficient to compose cordierite and an oxygen vacancy in the
35 cordierite crystal unit is formed. When the cordierite materials used are oxides only, the oxygen composing

cordierite can be completely supplied from the cordierite materials and, therefore, a reduced pressure atmosphere or a reducing atmosphere should be used to decrease the oxygen.

5 In the method (ii) of forming a cordierite honeycomb structure having oxygen vacancies, at least some Si, Al and Mg sources as the cordierite materials are replaced by a compound containing at least one of Si, Al and Mg and not containing oxygen. Such compounds may be
10 nitrides, halogenides such as fluorides and chlorides, and the like, which contain at least one of Si, Al and Mg elements constituting cordierite. Some or all of at least one of Si, Al and Mg sources are replaced by the above compounds which do not contain oxygen. The other
15 cordierite materials may be the same as those used in the method (i).

The above cordierite materials are formulated as to form a theoretical composition as described before and are formed into a honeycomb shape and heated to remove an organic binder, in the manner similar to the method (1), followed by firing in a low oxygen concentration atmosphere. The oxygen concentration of the atmosphere is from 0% to less than 3%, preferably from 0% to 1%. In this process, an oxygen vacancy is formed in the
25 cordierite crystal lattice. When a compound not containing oxygen is used in the cordierite materials, the oxygen contained in the cordierite materials is insufficient to compose cordierite and the firing atmosphere does not provide a sufficient amount of oxygen
30 to form a cordierite crystal by reaction since the concentration of oxygen in the firing atmosphere is low. Thus, an oxygen vacancy is formed in a cordierite crystal lattice.

When a compound not containing oxygen is used in the cordierite materials, a reduced pressure atmosphere or a reducing atmosphere, as used in the method (i), may be used in place of the low oxygen concentration atmosphere.

In such a case, a sufficient amount of oxygen is not supplied to compose a cordierite crystal during the reaction process and an oxygen vacancy is formed in the cordierite crystal lattice.

5 In the method (iii), at least one other-than-oxygen element constituting cordierite is replaced by an element having a valency smaller than the replaced element to form oxygen vacancies. In this method, the cordierite materials in which some Si, Al and Mg sources are
10 replaced by a compound containing an element having a valency smaller than that of the replaced Si, Al or Mg are used. The cordierite constituting elements, Si, Al and Mg, have valences of +4 (Si), +3 (Al) and +2 (Mg), respectively. Some of at least one of these elements may
15 be replaced by a compound having a valency smaller than the replaced element. Such compounds may be any of oxides, hydroxides, nitrides, halogenides and the like and the other cordierite materials may be usual cordierite materials to prepare the cordierite materials.
20 The prepared cordierite materials are formed into a honeycomb shape, heated to remove the organic binder, and then fired. The firing atmosphere may be any of a reduced pressure atmosphere, a reducing atmosphere, an oxygen-containing atmosphere such as an air atmosphere,
25 or an atmosphere not containing oxygen. Since the oxygen necessary to compose cordierite is contained in the cordierite materials and the oxygen vacancies are formed by element replacement, the oxygen concentration in the firing atmosphere does not affect the product and any
30 oxygen concentration in a range of 0 to 100% results in forming oxygen vacancies.

35 The cordierite constituting elements Si, Al and Mg have positive charges of +4 (Si), +3 (Al) and +2 (Mg). When any of the constituting elements is replaced by an element having a valency smaller than the replaced element, a positive charge corresponding to the difference of the replacing element with the replaced

element is deficient and, in order to maintain the electrical neutrality, the negatively charged oxygen (O^{2-}) is released. Thus, by replacing the cordierite-constituting element by an element having a smaller valency, an oxygen vacancy can be formed in the cordierite crystal unit.

In the method (iv), at least one other-than-oxygen element constituting cordierite is replaced by an element having a valency larger than the replaced element to form lattice defects. In this method, the cordierite materials in which some of Si, Al and Mg sources is replaced by a compound containing an element having a valency larger than that of the replaced Si, Al or Mg are used. Also, in this case, some of at least one of the cordierite-constituting elements may be replaced by a compound having a valency larger than the replaced element and the other cordierite materials may be usual cordierite materials to prepare the cordierite materials. The prepared cordierite materials are formed into a honeycomb shape, heated to remove the organic binder, and then fired. The firing atmosphere in the method (iv) should be an atmosphere which supplies a sufficient amount of oxygen, such as in air. When an air atmosphere is used as the firing atmosphere, the heating step for removing an organic binder may be eliminated since an organic binder can be removed during the firing.

When any of the constituting elements is replaced by an element having a valency larger than the replaced element, a positive charge corresponding to the difference of the replacing element with the replaced element is in excess and, in order to maintain the electrical neutrality, the negatively charged oxygen (O^{2-}) is incorporated in a necessary amount. The incorporated oxygen impedes arrangement of the cordierite crystal lattice in the regular manner, so that lattice defects are formed.

When an oxygen vacancy is formed in the unit lattice

of cordierite crystal, the amount of oxygen contained in the unit lattice of cordierite crystal becomes less than in the unit lattice of cordierite crystal without an oxygen vacancy. Also, the crystal lattice is deformed as the oxygen vacant portion is compressed, by which the lattice constant of b_0 axis of the cordierite crystal becomes smaller. On the other hand, when a lattice defect is formed in the unit lattice of cordierite crystal, the amount of oxygen contained in the unit lattice of cordierite crystal having such a lattice defect becomes more than in the unit lattice of cordierite crystal having no oxygen vacancy, and the lattice constant of b_0 axis of the cordierite crystal is changed. More specifically, when an oxygen vacancy is formed and the content of oxygen in the honeycomb structure becomes less than 47% by weight, the amount of oxygen contained in the unit cordierite crystal lattice becomes less than 17.2 and the lattice constant of b_0 axis of the cordierite crystal becomes smaller than 16.99. Also, when lattice defect is formed and the content of oxygen in the honeycomb structure becomes more than 48% by weight, the amount of oxygen contained in the unit lattice of cordierite crystal becomes more than 17.6 and the lattice constant of b_0 axis of the cordierite crystal becomes larger than 16.99.

As described above, in the present invention, due to oxygen vacancies or lattice defects formed in the cordierite crystal lattice, a required amount of a catalyst component can be supported on a cordierite honeycomb structure. Here, since the size of the defects are considered to be in the order of several angstroms or less, the specific surface area cannot be measured by usual specific surface area measuring methods such as the BET method using nitrogen atoms.

Next, a cordierite honeycomb structure having a large number of fine cracks in at least one of the amorphous and the crystal phase in the above method (2)

is described. The fine cracks are formed by (i) applying thermal shock or (ii) applying a shock wave, to a cordierite honeycomb structure, by which the fine cracks are formed in at least one of the amorphous and the crystal phase and the cordierite honeycomb structure is enabled to support a catalyst component. The fine cracks should have a width equivalent to or larger than the ion diameter of catalyst component, usually 0.1 nm or more, and a depth of a half or more of the ion diameter of catalyst component, usually 0.05 nm or more, to support the catalyst component. In order to maintain the strength of the honeycomb structure, smaller cracks are preferred, usually the width thereof being about 100 nm or less, more preferably about 10 nm or less.

The method (i) of applying thermal shock is performed by heating a cordierite honeycomb structure followed by rapid cooling. The application of thermal shock is carried out after a cordierite crystal phase and an amorphous phase. The application of thermal shock may be performed, either, by preparing cordierite materials, forming the cordierite materials into a shape, heating the shape to remove an organic binder and then firing the shape to form a cordierite honeycomb structure, as usual, followed by re-heating the cordierite honeycomb structure to a predetermined temperature and then rapidly cooling the heated cordierite honeycomb structure; or by rapidly cooling the cordierite honeycomb structure produced in the usual manner, during the cooling step following the firing step. Formation of fine cracks usually occurs if the temperature difference between before and after the rapid cooling (thermal shock temperature difference) is about 80° or more and the size of the cracks increases along with an increase in the size of the cracks. However, the thermal shock temperature difference is preferably not higher than about 900°C since, if the size of the cracks is too large, it is difficult to maintain

the shape of the honeycomb structure.

In a cordierite honeycomb structure, the amorphous phase is present in the form of layers around the crystal phase. If thermal shock is applied to a cordierite honeycomb structure by rapid cooling after heating the honeycomb structure, a thermal stress corresponding to a thermal shock temperature difference and a thermal expansion coefficient difference between the amorphous and crystal phases occurs near the interface between the amorphous and crystal phases. If the amorphous or crystal phase cannot endure the thermal shock, fine cracks appear. The amount of the fine cracks can be controlled by the amount of the amorphous phase present in the cordierite honeycomb structure. Since the fine cracks are formed near the amorphous phase, the amount of the fine cracks increases if the amount of the amorphous phase increases.

The amorphous phase in a cordierite honeycomb structure is considered to be formed by an alkali metal element or an alkaline earth metal element, which is contained in a small amount in the cordierite materials, which functions as a flux during the firing of the honeycomb structure. Therefore, the amount of fine cracks to be formed by thermal shock can be increased by adding an alkali metal element or an alkaline earth metal element to increase the amount of the amorphous phase. Further, by controlling the amount of the added alkali metal element or alkaline earth metal element, the amount of the fine cracks can be controlled. The effect of addition of an alkali metal element or alkaline earth metal element can be obtained when the alkali metal element or alkaline earth metal element is added in an amount of more than such an amount of the alkali metal elements and alkaline earth metal elements as impurities contained the cordierite materials, and, usually, more than 0.05% by weight of the total of the alkali metal element and alkaline earth metal element is sufficient.

Such an alkali metal element or alkaline earth metal element may be added as a compound containing an alkali metal element or alkaline earth metal element, for example, as an oxide, hydroxide, carbonate, etc. thereof in the step of preparing the cordierite materials.

In place of the thermal shock, the method (ii) of applying a shock wave may be used to introduce fine cracks in an amorphous or crystal phase. In this case, when the energy of the shock wave exceeds the strength of weak portions of a honeycomb structure, fine cracks are formed. The means for providing a shock wave include ultrasound wave, vibration, etc. and the amount of the formed fine cracks can be controlled by the energy, etc. of the shock wave.

It is possible that fine cracks may be additionally formed to at least one of amorphous and crystal phases of a cordierite honeycomb structure by the method (2), in which oxygen vacancies or lattice defects have been formed in cordierite crystal lattice by the method (1). In this case, after a cordierite honeycomb structure having oxygen vacancies or lattice defects and having an oxygen content of less than 47% by weight or more than 48% by weight and a lattice constant of crystal axis b_0 of larger or smaller than 16.99 is formed by firing in the method (1), a thermal shock or shock wave is applied to the honeycomb structure by the method (2) to obtain a cordierite honeycomb structure having at least ones of oxygen vacancies or lattice defects as well as a large number of fine cracks. To support a required amount of a catalyst component, it is sufficient to have a total number of the oxygen vacancies, lattice defects and fine cracks of $1 \times 10^7/l$ or more, preferably $1 \times 10^8/l$ or more. The application of a shock wave of method (2) can induce fine cracks in amorphous and/or crystal phases.

Next, the method (3), a liquid phase process, of dissolving constituting elements or impurities of cordierite to form vacancies in a cordierite honeycomb

structure, is now described. The vacancies are formed by dissolving a metal element such as Mg or Al in the cordierite crystal, an alkali metal element or alkaline earth metal element contained in the amorphous phase, or an amorphous phase itself, into a high temperature and high pressure water, a super critical fluid, an alkali solution or the like. A catalyst component can be supported on the fine pores formed by the above vacancies of elements, etc.

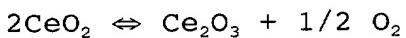
A cordierite honeycomb structure is first formed by a usual process, i.e., by preparing cordierite materials containing S, Al and Mg sources, forming the cordierite materials into a shape, heating the shape to remove an organic binder, and firing the shape. The thus produced cordierite honeycomb structure is then immersed in a high temperature and high pressure water, a super critical fluid or an alkali solution. As a result, metal elements such as Mg or Al in the cordierite crystal, alkali metal element and alkaline earth metal elements contained in the amorphous phase, or the amorphous phase itself are dissolved to form fine pores. The size of the fine pores can be controlled by the temperature of the solution, the pressure, the solvent, etc. Specifically, high temperature and high pressure water, at 10 MPa and 300°C, a super critical fluid of CO₂ or the like, and a solution such as an alkali solution, for example, sodium hydroxide, are used. By adding an alkali metal element or an alkaline earth metal element to cordierite materials, it is possible to form an amorphous phase, by which the amount of the fine pores can be also controlled.

The method (4), a vapor phase method, of chemically or physically forming vacancies in a cordierite honeycomb structure is described. Fine pores are formed by dry etching or sputter etching a cordierite honeycomb structure. In the dry etching, a reaction gas is excited by discharging the reaction gas by radio frequency energy

or others. The reaction gas reacts with Si, Al and/or Mg of the cordierite-constituting elements to form volatile substances, which are volatilized and evacuated, by which the cordierite is etched. The thus chemically etched
5 vacancies constitute fine pores on which a catalyst component can be supported. The reaction gas may be CF₄, etc., which reacts with the cordierite-constituting elements to form volatile substances such as Si₄F, etc. The level of the dry etching can be controlled by the
10 etching time period, the kind of the reaction gas, the supplied energy, etc.

In the sputter etching, a cordierite honeycomb structure is placed in a plasma of Ar, etc., which is excited by radio frequency wave, etc., by which the Ar ions, etc. bomb the surface of the cordierite to sputter atoms or masses of pluralities of atoms of the cordierite-constituting elements, thus sputter etching the cordierite. The thus physically etched vacancies constitute fine pores where a catalyst component can be supported. The level of the sputter etching can be controlled by the etching time period, the kind of the excited gas, the supplied energy, etc.
15
20

Next, the method (5) of incorporating an oxygen storage substance in a cordierite honeycomb structure is described. A substance capable of storing oxygen, for example, CeO₂, incorporates and releases oxygen in the reversible reaction represented by the following formula:
25



in accordance with the change in the oxygen concentration in the atmosphere. That is, when the oxygen concentration in the atmosphere is high, the valence of Ce is +4, but when the oxygen concentration in the atmosphere is low, the valence of Ce becomes +3, by which, due to unbalanced electrical neutrality caused by the change in the valency, the oxygen storage substance releases or incorporates oxygen to maintain the electrical neutrality. Such an oxygen storage substance
30
35

is used as a co-catalyst in a three way converter catalyst in which oxygen is incorporated and released depending on the change in the concentration of oxygen in an exhaust gas to control the air-to-fuel ratio to near
5 the theoretical air-to-fuel ratio.

If cerium, Ce, which can take a plurality of valencies, is substituted for a cordierite-constituting element in a cordierite honeycomb structure, in order to complement the change of the valency, as in the
10 method (1), an excess or deficiency of oxygen is caused to form an oxygen vacancy or lattice defect in the cordierite crystal lattice. The oxygen vacancy or lattice defect results in fine pores allowing catalyst support and providing the cordierite honeycomb structure
15 with the oxygen storage capacity. That is, without a coating of γ -alumina, a catalyst component can be directly supported and, without separately supporting a co-catalyst, an oxygen storage capacity is provided. In order to provide the oxygen storage capacity, the content
20 of CeO_2 in a cordierite honeycomb structure is desirably made in a range of 0.01% by weight or less.
25

In order to obtain a cordierite honeycomb structure containing CeO_2 , Ce is substituted for some of at least one of Si, Al and Mg elements constituting the cordierite. The substitution may be in the same manner as in the method (1), and can be made by using cordierite materials in which a compound containing Ce in place of Si, Al or Mg is substituted for some of the Si, Al and Mg sources. Since Ce has a valence of +4 in air, if Ce is substituted for Mg (+2) or Al (+3) having a smaller
30 valence, a lattice defect is formed as in the similar manner as in the above (iv), and even if Ce is substituted for Si (+4), an oxygen vacancy is formed since some of Ce has a valence of +3.

35 Thus, by using Ce as a substituting element, a cordierite honeycomb structure having a catalyst supporting capability and an oxygen storage capacity can

be obtained. If CeO₂ is supported as a co-catalyst, CeO₂ may be thermally deteriorated and result in grain growth, lowering the oxygen storage capacity. However, if CeO₂ is incorporated in the cordierite structure, no grain growth occurs and the oxygen storage capacity is not deteriorated.

After the cordierite honeycomb structure is fired, a thermal shock or shock wave may be applied as in the above method (2) to form fine cracks. This increases the number of the fine pores improving the catalyst supporting capability. Alternatively, in combination with the method, elements other than Ce may be used, or the firing atmosphere may be controlled, to control the number of oxygen vacancies and lattice defects formed.

A co-catalyst having an oxygen storage capacity such as CeO₂ may be additionally supported on a cordierite honeycomb structure in which the catalyst supporting capability has been provided by the above methods (1) to (4). In this case, since the co-catalyst can be supported without coating γ -alumina and utilizing fine pores of the catalyst supporting capability, a cordierite honeycomb structure having a catalyst supporting capability and an oxygen storage capacity can be easily obtained. Supporting a co-catalyst having an oxygen storage capacity may be done by supporting a precursor of a co-catalyst such as an ion or a complex thereof followed by heat treatment.

The cordierite honeycomb structure having a catalyst supporting capability, as produced by the above methods, may be preferably used as a ceramic support used for a catalyst for cleaning an exhaust gas of an internal combustion gas. In this ceramic support, 0.1 g/l or more of a catalyst component can be supported on the fine pores of the cordierite honeycomb structure without a coating of γ -alumina, so that a catalyst-ceramic body having a low thermal capacity, a high thermal shock

resistance and a low pressure loss can be obtained. The catalyst metal component may be at least one of metals having a catalyst activity and metal oxides having a catalyst activity. The metals having a catalyst activity include noble metals such as Pt, Pd and Ph and the metal oxides having a catalyst activity include oxides containing at least one metal of V, Nb, Ta, Cr, Mo, W, Mn, Fe, Co, Ni, Cu, Zn, Ga, Sn, Pb, etc.

The methods for supporting a catalyst component include the liquid phase methods, in which a catalyst component is dissolved in a solvent and impregnated into a cordierite honeycomb structure to support the catalyst component in the fine pores such as defects and cracks, and the vapor deposition methods such as CVD and PVD, as well as other methods including use of a super critical fluid. It is preferred to use a vapor phase deposition methods or a method using a solvent such as a super critical fluid which can be infiltrated into deep inside the fine pores, since the fine pores such as defects or cracks formed in the cordierite honeycomb structure are very fine. In the liquid phase methods, although water may be used as the solvent, a solvent having a smaller surface tension than water, for example, an alcohol solvent such as methanol is preferred. By using a solvent having a smaller surface tension than water, the solution can be sufficiently infiltrated into the fine pores. In this case, if the infiltration is performed while vibration is applied or vacuum defoaming is carried out, the solvent can be more easily introduced into the fine pores. Further, a catalyst component may be supported on a ceramic support by a plurality of supporting steps using the same or different compositions until a necessary amount of the catalyst component is supported. These methods may be effectively used to support a catalyst component in an amount of 0.5 g/l or more.

The catalyst-ceramic body of the present invention

obtained by the above process is a catalyst-ceramic body excellent in cleaning performance, which supports a required amount of a catalyst component directly thereon with a narrow space therebetween, without a coating of
5 γ -alumina. Specifically, a catalyst component may be supported until the content of the metal element in the catalyst-ceramic body supporting the catalyst component, becomes 0.01% by weight, and the average space between the catalyst component particles (particles of catalyst
10 component ion and catalyst metal, etc.) supported on the catalyst-ceramic body is as small as 0.1 to 100 nm. This means that with the same supporting amount of a catalyst component, the particle size of the catalyst component particles become smaller and the number thereof larger,
15 which results in a denser distribution of the catalyst component on the entire surface of the ceramic support and the catalyst exhibits its performance more effectively.

Moreover, in the present invention, a metal having a catalyst captivity may be substituted for a constitutional element of a cordierite honeycomb structure, to obtain a catalyst-ceramic body. In this case, cordierite materials in which some the Si, Al and Mg sources are replaced by a compound containing a metal having a catalyst activity, preferably a noble metal such as Pt, Pd or Rd, in place of Si, Al or Mg of the cordierite-constituting elements, are used. Such cordierite materials are formed into a shape, heated to remove an organic binder and then fired to form a ceramic catalyst body. The firing atmosphere may be a reducing pressure atmosphere having a pressure of 4000 MPa or less, a reducing atmosphere such as a hydrogen atmosphere, or an oxygen-containing or an oxygen-free atmosphere. If a metal having an oxygen storage capacity such as Ce is used also as a substituting metal, in addition to a metal having a catalyst activity, the catalyst-ceramic body can support a catalyst component as

well as a co-catalyst having an oxygen storage capacity.

BRIEF DESCRIPTION OF THE DRAWING

Fig. 1 shows a cordierite honeycomb structure
5 wherein the reference numeral 1 denotes a cordierite
honeycomb structure, 2 cells of flow channel, and 3 the
direction of flow channel.

EXAMPLES

10 (Example 1)

In Example 1, as the cordierite materials, talc,
kaolin, alumina and aluminium hydroxide were used and
formulated into a near theoretical cordierite
composition. The cordierite materials were added with
15 appropriate amounts of a binder, a lubricant and a
humidicant, kneaded to a clay-like material and extruded
to form a honeycomb shape having a cell wall thickness of
100 μm , a cell density of 400 cpsi (cells per square
inch) and a diameter of 50 mm. The honeycomb shape was
20 dried by heating, in air, up to 800°C to remove the
binder, and then fired by reducing the pressure to
 6.7×10^{-4} Pa (5×10^{-6} Torr) and keeping the honeycomb
shape under that pressure at 1390°C for 2 hours, to
obtain a honeycomb structure.

25 The obtained honeycomb structure was evaluated in
the oxygen content of the honeycomb structure, the number
of oxygen contained in the unit lattice of the cordierite
crystal, the lattice constant of the b_0 axis of the
cordierite crystal, the amount of Pt supported on the
30 honeycomb structure (see below), the thermal expansion
coefficient in the direction of the flow channel and the
compressive strength in the direction of the flow
channel. The results are shown in Table 1.

35 The oxygen content of the honeycomb structure was
measured by a simultaneous oxygen and nitrogen analyser.
The number of oxygen contained in the unit lattice of the

cordierite crystal was calculated from the amount of oxygen contained in the honeycomb structure. The lattice constant of the b_0 axis of the cordierite crystal was obtained from the positions of diffraction peaks of (020) plane of the cordierite unit crystals by the powder X-ray diffraction, using the (112+) plane of Mn_2O_3 as the reference by adding Mn_2O_3 to the measured sample for correction of the positions of the diffraction peaks.

The amount of Pt supported on the honeycomb structure was measured using X-ray fluorescence spectroscopy by supporting a catalyst component on the honeycomb structure and pulverizing the honeycomb structure supporting the catalyst component. The supporting of Pt as the catalyst component was done by preparing a solution of chloroplatinic acid in water or ethanol in a concentration of 0.1 mol/l, immersing the honeycomb structure in the solution, drying, and firing or heating the immersed honeycomb structure in air at 800°C for 2 hours.

The thermal expansion coefficient in the direction of the flow channel was measured using a pressing bar-type thermal expansion meter as the average thermal expansion coefficient from 25°C to 800°C. The compressive strength in the direction of the flow channel was determined by cutting out a cylinder of the honeycomb structure with a diameter of 1 inch and a length of 1 inch and applying a load to the cylinder in the direction of the flow channel until the cylinder was broken, the pressure at breaking being used as the compressive strength.

For comparison, a cordierite honeycomb structure was produced in the same manner as above except that the honeycomb shape was fired in air at 1390°C for 2 hours, which is Comparative Example 1. This cordierite honeycomb structure was evaluated similarly and the results are also shown in Table 1.

In comparison with Comparative Example 1, the cordierite honeycomb structure fired in a reduced pressure atmosphere had a reduced content of the honeycomb structure, a reduced amount of oxygen contained 5 in the unit lattice of the cordierite crystal, and a smaller lattice constant of the b_0 axis of the cordierite crystal. It is clear that oxygen left the cordierite crystals and oxygen vacancies were formed.

With respect to the amount of the supported Pt, only 10 a very small amount of Pt, too small to be detected, was supported on the cordierite honeycomb structure in both the cases where water and ethanol were used as the solvent in Comparative Example 1. In Example 1, in contrast, Pt was supported in an amount of 1.00 g/l in 15 the case of water and 13.79 g/l in the case of ethanol. It is clearly shown that Pt can be supported in a sufficient amount on the honeycomb structure of Example 1. Further, the reason why the amount of the supported Pt is lower in the case of water as the solvent 20 is because, with a solvent having such high surface tension as water, the Pt solution cannot come close to the oxygen, and vacancies and Pt ions are difficult to incorporate into the oxygen vacancies.

Further, in Example 1, the thermal expansion 25 coefficient in the direction of the flow channel was $0.92 \times 10^{-6}/^{\circ}\text{C}$, satisfying the requirement of not more than $1.0 \times 10^{-6}/^{\circ}\text{C}$ required for a catalyst support, and the compressive strength in the direction of the flow channel was 11.92 MPa, exceeding the 10 MPa required for 30 a honeycomb structure to endure a load for assembling into a catalyst converter.

The oxygen vacancies of the honeycomb structure of Example 1 were observed by Transmission Electron Microscopy (TEM). The size of the oxygen vacancies was 35 about 0.4 nm. Since this size is about four times the

diameter of the Pt ion, about 0.1 nm, these oxygen vacancies are sufficiently large to support Pt ions.

The amount of the oxygen vacancies of the honeycomb structure in Example 1 can be calculated as below. The amount of the oxygen vacancies of the honeycomb structure in Example 1 corresponds to the difference of the content of oxygen between the honeycomb structure produced in Comparative Example 1 and the honeycomb structure in Example 1. Since the above difference in the contents was 2.0% by weight, this corresponds to 0.7 oxygen contained in the unit lattice of the cordierite crystal. Since the number of the unit lattices of the cordierite crystal contained in the cordierite honeycomb structure is $2.36 \times 10^{23}/l$, the number of oxygen vacancies becomes $1.65 \times 10^{23}/l$ by the following formula:

$$(2.36 \times 10^{23}/l) \times 0.7 = 1.65 \times 10^{23}/l$$

This number of oxygen vacancies far exceeds the number of fine pores required for a catalyst support, as mentioned before, i.e., $1 \times 10^{16}/l$, preferably $1 \times 10^{17}/l$.

In contrast, in Comparative Example 1, the cordierite honeycomb structure had a large number of cracks, so called microcracks, having a size of about several hundreds nanometers. However, the amount of Pt supported on the cordierite honeycomb structure in Comparative Example 1 was in such a small amount that the supported Pt was not detected, as mentioned above. It is considered that Pt was not supported on microcracks, since the width of the microcracks is about a few thousands times the diameter of Pt ion and is too large to support Pt therein in the supporting step.

Table 1

Sample	Firing atmosphere	Oxygen content (wt%)	Number of oxygen	Lattice constant b_0 (Å)	Pt supported amount (g/L) Aqueous solution	Thermal expansion coefficient ($\times 10^{-6}/^{\circ}\text{C}$) Ethanol solution	Compressive strength (MPa)	Remarks
Comparative Example 1	Air atmosphere	47.0	17.2	16.99	0	0	0.47	18.0
Example 1	Reduced pressure atmosphere 6.7×10^{-4} Pa	45.0	16.5	16.96	1.0	13.8	0.92	11.9
Example 2	Hydrogen atmosphere	42.5	15.5	16.94	1.1	14.8	0.99	10.2

(Example 2)

The same cordierite materials as in Example 1 was formed into a honeycomb shape, heated to remove a binder and then fired in a hydrogen atmosphere as a reducing atmosphere at 1390°C for 2 hours.

5 The obtained honeycomb structure was evaluated in the same manner as in Example 1 and the results are also shown in Table 1.

10 It is clear from Table 1 that in comparison with Comparative Example 1, the cordierite honeycomb structure fired in a reducing atmosphere in Example 1 had a reduced content of the honeycomb structure, a reduced amount of oxygen contained in the unit lattice of the cordierite crystal, and a smaller lattice constant of the b_0 axis of 15 the cordierite crystal. It is therefore clear that oxygen left the cordierite crystals and oxygen vacancies were formed. When Pt was supported on the cordierite honeycomb structure, the amount of the Pt supported was 20 1.14 g/l in the case of water used as the solvent and 14.81 g/l in the case of ethanol used as the solvent. The thermal expansion coefficient in the direction of the flow channel was $0.99 \times 10^{-6}/^\circ\text{C}$, and the compressive strength in the direction of the flow channel was 10.2 MPa, both of which satisfy the requirements.

25 The amount of the oxygen vacancies of the honeycomb structure in Example 2 was also calculated. The amount of the oxygen vacancies of the honeycomb structure in Example 2 corresponds to the difference of the content of oxygen between the honeycomb structure produced in 30 Comparative Example 1 and the honeycomb structure in Example 2. Since this difference of the content was 4.5% by weight, this corresponds to 1.7 oxygens contained in the unit lattice of the cordierite crystal. Since the number of the unit lattices of the cordierite crystal 35 contained in the cordierite honeycomb structure is

$2.36 \times 10^{23}/l$, the number of oxygen vacancies becomes $4.01 \times 10^{23}/l$ by the following formula:

$$(2.36 \times 10^{23}/l) \times 1.7 = 4.01 \times 10^{23}/l$$

Thus, in accordance with the present invention,
5 there is provided a cordierite honeycomb structure in which fine pores having a size suitable to support a catalyst component are formed in a sufficient number for supporting the catalyst component.

(Examples 3 to 6 and Comparative Examples 2 to 7)

10 In Example 3, Si_3N_4 was used for 10% by weight of the Si source and, talc, kaolin, alumina and aluminum hydroxide were used as the other Si, Al and Mg sources to formulate a near theoretical cordierite composition. The thus prepared cordierite materials were formed into a
15 honeycomb shape, heated to remove a binder and then fired or heated at 1390°C in an atmosphere having an oxygen concentration of 1% for 2 hours to produce a cordierite honeycomb structure.

20 In Example 4, AlF_3 was used for 10% by weight of the Al source and talk, kaolin, alumina and aluminum hydroxide were used as the other Si, Al and Mg sources to formulate a near theoretical cordierite composition. Using these cordierite materials, a cordierite honeycomb structure was produced in the same manner as above.

25 In Example 5, MgCl_2 was used for 10% by weight of the Mg source and talc, kaolin, alumina and aluminum hydroxide were used as the other Si, Al and Mg sources to formulate a near theoretical cordierite composition. Using these cordierite materials, a cordierite honeycomb structure was produced in the same manner as above.

30 The obtained cordierite honeycomb structures were evaluated in the same manner as in Example 1. The results are shown in Table 2.

35 For comparison, cordierite honeycomb shapes were made using the same cordierite materials as in Examples 3 to 5 and were fired in an atmosphere having an oxygen

content of 3% or 12%. These were made as Comparative Examples 2 to 7. The obtained cordierite honeycomb structures were evaluated in the same manner as in Example 1. The results are shown in Table 2.

5 As seen in Table 2, in comparison with Comparative Examples 2 to 7, the cordierite honeycomb structures produced by firing in an atmosphere having an oxygen concentration of 1% in Examples 3 to 5 had a reduced oxygen content of the honeycomb structure, a reduced
10 number of the oxygen contained in the unit lattice of the cordierite crystal, and a smaller lattice constant of the b_0 axis of the cordierite crystal. It is therefore clear that oxygen vacancies were formed.

15 When Pt was supported on the cordierite honeycomb structures of Examples 3 to 5, Pt was supported in an amount of 0.21 to 0.35 g/l in the case of water as the solvent and in an amount of 2.88 to 3.44 g/l in the case of ethanol. The thermal expansion coefficient in the direction of the flow channel was 0.89 to $0.94 \times 10^{-6}/^{\circ}\text{C}$,
20 and the compressive strength in the direction of the flow channel was 12.8 to 14.3 MPa, both of which satisfy the requirements.

Table 2

Sample	Replacing cordierite material	Firing atmosphere	Oxygen content (wt%)	Number of oxygen b ₀	Lattice constant (Å)	Pt supported amount (g/L) Aqueous solution	Ethanol solution	Thermal expansion coefficient ($\times 10^{-6}/^{\circ}\text{C}$)	Compressive strength (MPa)	Remarks
Example 3	Si ₃ N ₄	Oxygen concentration: 1%	46.0	16.8	16.98	0.35	3.04	0.89	13.3	
Example 4	AlF ₃	Oxygen concentration: 1%	46.0	16.8	16.98	0.24	3.44	0.94	14.3	
Example 5	MgCl ₂	Oxygen concentration: 1%	46.5	17.0	16.98	0.21	2.88	0.91	12.8	
Comparative Example 2	Si ₃ N ₄	Oxygen concentration: 3%	47.5	17.4	16.99	0	0	0.39	17.9	
Comparative Example 3	AlF ₃	Oxygen concentration: 3%	47.0	17.2	16.99	0	0	0.86	17.6	
Comparative Example 4	MgCl ₂	Oxygen concentration: 3%	48.0	17.6	16.99	0	0	0.83	17.5	
Comparative Example 5	Si ₃ N ₄	Oxygen concentration: 12%	47.0	17.2	16.99	0	0	0.32	18.1	
Comparative Example 6	AlF ₃	Oxygen concentration: 12%	47.5	17.4	16.99	0	0	0.80	18.3	
Comparative Example 7	MgCl ₂	Oxygen concentration: 12%	47.5	17.4	16.99	0	0	0.75	17.8	
Example 6	Si ₃ N ₄	Reduced pressure atmosphere 6.7 × 10 ⁻⁴ Pa	46.0	16.8	16.96	1.80	5.56	0.85	11.0	
Example 7	AlF ₃	Reduced pressure atmosphere 6.7 × 10 ⁻⁴ Pa	45.5	16.6	16.93	3.10	14.62	0.97	15.1	
Example 8	MgCl ₂	Reduced pressure atmosphere 6.7 × 10 ⁻⁴ Pa	46.5	17.0	16.96	1.03	7.94	0.97	12.2	
Example 9	AlF ₃	Hydrogen atmosphere	43.0	15.7	16.94	1.01	12.14	0.96	10.9	
Example 10	MgCl ₂	Hydrogen atmosphere	44.5	16.3	16.96	1.26	10.91	0.98	10.7	

(Examples 6 to 10)

In Examples 6 to 8, cordierite honeycomb structures were produced in the same manner as in Examples 3 to 5, except that the used firing atmosphere was a reduced pressure atmosphere having a pressure of $6.7 \text{ to } 10^{-4} \text{ Pa}$ (5 $\times 10^{-6}$ Torr).

In Examples 9 and 10, cordierite honeycomb structures were produced in the same manner as in Example 5, except that the used firing atmosphere was a hydrogen atmosphere as a reducing atmosphere.

The obtained cordierite honeycomb structures were evaluated in the same manner as in Example 1. The results are also shown in Table 2.

As seen in Table 2, in comparison with Comparative Examples 2 to 7, the cordierite honeycomb structures produced in Examples 6 to 10 had a reduced oxygen content of the honeycomb structure, a reduced number of the oxygen contained in the unit lattice of the cordierite crystal, and a smaller lattice constant of the b_0 axis of the cordierite crystal. It is therefore clear that oxygen vacancies were formed. When Pt was supported, Pt was supported in an amount of 1.01 to 3.10 g/l in the case of water as the solvent and in an amount of 5.56 to 14.62 g/l in the case of ethanol. The thermal expansion coefficient in the direction of the flow channel was 0.85 to $0.98 \times 10^{-6}/^\circ\text{C}$, and the compressive strength in the direction of the flow channel was 10.7 to 15.1 MPa, both of which satisfy the requirements.

(Examples 11 to 18)

In Examples 11 and 12, Fe_2O_3 or Ca_2O_3 was used for 10% by weight of the Si source, and talc, kaolin, alumina and aluminum hydroxide were used as the other Si, Al and Mg sources to formulate a near theoretical cordierite composition. The thus prepared cordierite materials were formed into a honeycomb shape, heated to remove a binder

and then fired at 1390°C in an air atmosphere for 2 hours to produce a cordierite honeycomb structure.

The obtained cordierite honeycomb structures were evaluated in the same manner as before. The results are shown in Table 3.

In Examples 13 and 14, GeO_2 or MoO_3 , as an oxide or an element having a valence larger than Al, was used for 10% by weight of the Al source, and talc, kaolin, alumina and aluminum hydroxide were used as the other Si, Al and Mg sources to formulate a near theoretical cordierite composition. The thus prepared cordierite materials were formed into a honeycomb shape, heated and then fired in the same manner as in Example 11.

In Examples 15 to 18, Fe_2O_3 , Ga_2O_3 , GeO_2 or MoO_3 , as an oxide of an element having a valence larger than Mg, was used for 10% by weight of the Mg source, and talc, kaolin, alumina and aluminum hydroxide were used as the other Si, Al and Mg sources to formulate a near theoretical cordierite composition. The thus prepared cordierite materials were formed into a honeycomb shape, heated and then fired in the same manner as in Example 11.

The obtained cordierite honeycomb structures were evaluated in the same manner as before. The results are also shown in Table 3.

In Table 3, in comparison with Comparative Example 1, the cordierite honeycomb structures of Examples 11 and 12, in which some Si was replaced by an element having a smaller valence than Si, had a reduced oxygen content of the honeycomb structure, a reduced number of the oxygen contained in the unit lattice of the cordierite crystal, and a smaller lattice constant of the b_0 axis of the cordierite crystal. It is therefore clear that oxygen vacancies were formed. When Pt was supported, Pt was supported in an amount of 0.15 to 0.23 g/l in the case of water as the solvent and in an amount of 2.62 to 5.31 g/l in the case of ethanol. The

thermal expansion coefficient in the direction of the flow channel was 0.86 to $0.89 \times 10^{-6}/^{\circ}\text{C}$, and the compressive strength in the direction of the flow channel was 12.5 to 16.7 MPa, both of which satisfy the requirements.

In comparison with Comparative Example 1, the cordierite honeycomb structures of Examples 13 to 18, in which some cordierite-constituting element Al or Mg was replaced by an element having a larger valence than Al or Mg, had an increased oxygen content of the honeycomb structure, an increased number of the oxygen contained in the unit lattice of the cordierite crystal, and a varied lattice constant of the b_0 axis of the cordierite crystal. It is therefore clear that lattice defects were formed. When Pt was supported, Pt was supported in an amount of 0.27 to 1.24 g/l in the case of water as the solvent and in an amount of 0.67 to 3.69 g/l in the case of ethanol. The thermal expansion coefficient in the direction of the flow channel was 0.57 to $0.95 \times 10^{-6}/^{\circ}\text{C}$, and the compressive strength in the direction of the flow channel was 10.8 to 14.8 MPa, both of which satisfy the requirements.

(Examples 19 to 22)

In Examples 19 and 20, cordierite honeycomb structures were produced in the same manner as in Examples 11 and 12, except that the used firing atmosphere was a reduced pressure atmosphere having a pressure of 6.7 to 10^{-4} Pa (5×10^{-6} Torr).

In Examples 21 and 22, cordierite honeycomb structures were produced in the same manner as in Examples 11 and 12, except that the used firing atmosphere was a hydrogen atmosphere as a reducing atmosphere.

The obtained cordierite honeycomb structures were evaluated in the same manner as in Example 1. The results are also shown in Table 3.

As seen in Table 3, in comparison with Comparative Example 1, the cordierite honeycomb structures produced in Examples 19 and 20 had a reduced oxygen content of the honeycomb structure, a reduced number of the oxygen contained in the unit lattice of the cordierite crystal, and a smaller lattice constant of the b_0 axis of the cordierite crystal. It is therefore clear that oxygen vacancies were formed. When Pt was supported, Pt was supported in an amount of 0.25 to 0.68 g/l in the case of water as the solvent and in an amount of 2.77 to 10.15 g/l in the case of ethanol. The thermal expansion coefficient in the direction of the flow channel was 0.91 to $0.98 \times 10^{-6}/^{\circ}\text{C}$, and the compressive strength in the direction of the flow channel was 10.0 to 11.0 MPa, both of which satisfy the requirements.

(Examples 23 and 24)

In Example 23, a cordierite honeycomb structure was produced in the same manner as in Example 11, except that the used firing atmosphere was a nitrogen atmosphere having an oxygen concentration of 0%.

In Example 24, a cordierite honeycomb structure was produced in the same manner as in Example 11, except that the used firing atmosphere was an oxygen atmosphere having an oxygen concentration of 100%.

The obtained cordierite honeycomb structures were evaluated in the same manner as before. The results are also shown in Table 3.

In comparison with the cordierite honeycomb structures in Comparative Example 1, the cordierite honeycomb structures prepared in Examples 23 and 24, where a portion of the cordierite constituting element Si was replaced by an element having a smaller valence than Si, had a reduced oxygen content of the honeycomb structure, a reduced amount of oxygen contained in the unit lattice of the cordierite crystal, and a smaller lattice constant of the b_0 axis of the cordierite crystal. It is therefore clear that oxygen vacancies

were formed in the cordierite crystals. Here, when Examples 23 and 24 are compared, the characteristics of the obtained cordierite honeycomb structures are not very different from each other, although the oxygen concentration was different in these Examples.

5 Therefore, it may be concluded that the oxygen concentration of an oxygen atmosphere does not have an effect on the characteristics of the cordierite honeycomb structure.

10 When Pt was supported on the cordierite honeycomb structure, the amount of Pt supported was 0.20 to 0.25 g/l in the case of water as the solvent and 3.01 to 4.98 g/l in the case of ethanol as the solvent. The thermal expansion coefficient was 0.85 to $0.90 \times 10^{-6}/^{\circ}\text{C}$ 15 and the compressive strength was 12.4 to 16.6 MPa, both of which satisfy the requirements.

Table 3

Sample	Specific cordierite material	Firing atmosphere	Oxygen content (wt%)	Number of oxygen	Lattice constant b_0 (Å)	Pt supported amount (g/L)	Thermal expansion coefficient ($\times 10^{-6}/^{\circ}\text{C}$)	Compressive strength (MPa)	Remarks
				Aqueous solution	Ethanol solution				
Example 11	Fe_2O_3	Air atmosphere	46.5	17.0	16.98	0.23	2.62	0.86	12.5 Si was replaced by Fe
Example 12	Ga_2O_3	Air atmosphere	45.5	16.6	16.98	0.15	5.31	0.89	16.7 Si was replaced by Ga
Example 13	GeO_2	Air atmosphere	48.5	17.7	17.01	1.24	1.80	0.74	14.1 Al was replaced by Ge
Example 14	MoO_3	Air atmosphere	48.5	17.7	16.98	0.69	0.67	0.80	11.9 Al was replaced by Mo
Example 15	Fe_2O_3	Air atmosphere	49.5	18.1	16.96	0.28	3.96	0.68	13.3 Mg was replaced by Fe
Example 16	Ga_2O_3	Air atmosphere	49.0	17.9	17.01	0.41	0.71	0.62	14.8 Mg was replaced by Ga
Example 17	GeO_2	Air atmosphere	48.5	17.7	17.02	0.90	2.04	0.57	12.2 Mg was replaced by Ge
Example 18	MoO_3	Air atmosphere	52.0	19.0	17.01	0.27	1.25	0.95	10.8 Mg was replaced by Mo
Example 19	Fe_2O_3	Reduced pressure atmosphere $6.7 \times 10^{-4} \text{ Pa}$	45.0	16.5	16.96	0.48	2.77	0.91	10.5 Si was replaced by Fe
Example 20	Ga_2O_3	Reduced pressure atmosphere $6.7 \times 10^{-4} \text{ Pa}$	44.5	16.3	16.96	0.25	6.29	0.93	11.0 Si was replaced by Ga
Example 21	Fe_2O_3	Hydrogen atmosphere	44.0	16.1	16.94	0.68	10.15	0.98	10.0 Si was replaced by Fe
Example 22	Ga_2O_3	Hydrogen atmosphere	43.0	15.7	16.94	0.56	8.69	0.95	10.6 Si was replaced by Ga
Example 23	Fe_2O_3	Nitrogen atmosphere (O_2 : 0%)	46.5	17.0	16.98	0.25	3.01	0.85	12.4 Si was replaced by Fe
Example 24	Ga_2O_3	Oxygen atmosphere (O_2 : 100%)	45.5	16.6	16.98	0.20	4.98	0.90	16.6 Si was replaced by Ga

(Examples 25 to 36)

As the cordierite materials, talc, kaolin, alumina and aluminium hydroxide were used and formulated into a near theoretical cordierite composition. In the same manner as in Example 1, the cordierite materials were added to a binder, a lubricant and a humidicant in appropriate amounts, formed into a honeycomb shape, and then fired in air at 1390°C for 2 hours, to obtain a honeycomb structure.

After the cordierite honeycomb structure was cooled to room temperature, the obtained cordierite honeycomb structure was subjected to a thermal shock by heating the honeycomb structure up to a temperature of the sum of the room temperature plus an intended thermal shock temperature difference followed by dropping the heated honeycomb structure into water. The thermal shock temperature difference was varied in a range of 150 to 900°C in Examples 25 to 30 as shown in Table 4.

Also, in Examples 31 to 36, in the course of cooling from the firing temperature of 1390°C above, when the cordierite honeycomb structure was cooled to a temperature of the sum of the room temperature plus an intended thermal shock temperature difference, air was blown to the honeycomb structure in the direction of the flow channel to provide a thermal shock to the honeycomb structure. The thermal shock temperature difference was varied in a range of 150 to 900°C in Examples 31 to 36 as shown in Table 4.

The honeycomb structures after the thermal shock were observed by TEM. It was confirmed that fine cracks having a width of 10 nm or less were formed in a large number at interfaces between the cordierite crystal phases and the amorphous phases.

The Pt supporting amount, the thermal expansion coefficient and the compressive strength in the direction

of the flow channel of the honeycomb structures were evaluated. The results are shown in Table 4.

As seen in Table 4, the Pt supporting amount was in a range of 0.88 to 1.64 g/l in the case of water as the solvent and in a range of 3.18 to 4.77 g/l in the case of ethanol, which indicates that supporting Pt was made possible by the fine cracks. The thermal expansion coefficient was 0.28 to $0.39 \times 10^{-6}/^{\circ}\text{C}$, and the compressive strength in the direction of the flow channel was 13.0 to 17.0 MPa, both of which satisfy the requirements.

Table 4

Sample	Temperature difference of thermal shock (°C)	Pt supported amount (g/L)		Thermal expansion coefficient ($\times 10^{-6}/^{\circ}\text{C}$)	Compressive strength (MPa)	Remarks
		Aqueous solution	Ethanol solution			
Example 25	150	1.35	3.43	0.29	16.8	Re-heating
Example 26	300	1.57	4.22	0.29	15.9	Re-heating
Example 27	450	0.88	3.18	0.32	15.1	Re-heating
Example 28	600	0.90	4.26	0.37	14.2	Re-heating
Example 29	750	1.44	4.61	0.37	13.6	Re-heating
Example 30	900	1.06	4.77	0.36	13.0	Re-heating
Example 31	150	0.98	3.33	0.29	17.0	During cooling
Example 32	300	1.64	4.13	0.28	16.1	During cooling
Example 33	450	1.33	3.76	0.34	15.3	During cooling
Example 34	600	1.20	4.01	0.38	14.8	During cooling
Example 35	750	1.03	4.35	0.37	14.0	During cooling
Example 36	900	1.10	4.44	0.39	13.2	During cooling

(Examples 37 to 39)

In Example 37, Ga_2O_3 , an oxide of an element having a smaller valence than Si, was used for 10% by weight of the Si source, and talc, kaolin, alumina and aluminum hydroxide were used as the other Si, Al and Mg sources to formulate a near theoretical cordierite composition. The thus prepared cordierite materials were formed into a honeycomb shape, heated to remove a binder and then fired at 1390°C in an air atmosphere for 2 hours to produce a cordierite honeycomb structure, in the same manner as in Example 1.

The obtained cordierite honeycomb structure was cooled to room temperature and re-heated to 320°C, at which temperature the re-heated honeycomb structure was dropped into water for rapid cooling or applying a thermal shock. Pt was supported onto the thus treated honeycomb structure by CVD. While a carrier gas of nitrogen was flowed at 20 l/min, platinum acetylacetone was heated and sublimated at 180°C, to adsorb Pt on the honeycomb structure for 1 hour and the honeycomb structure was then heat-treated in air at 800°C for 2 hours. The amount of Pt supported on the obtained honeycomb structure was measured to be 1.22 g/l.

In Examples 38 and 39, GeO_2 , an oxide of an element having a larger valence than Al, was used for 10% by weight of the Al source, and talc, kaolin, alumina and aluminum hydroxide were used as the other Si, Al and Mg sources to formulate a near theoretical cordierite composition. The thus prepared cordierite materials were formed into a honeycomb shape, heated to remove a binder and then fired at 1390°C in an air atmosphere for 2 hours to produce a cordierite honeycomb structure, in the same manner as in Example 1. In the course of cooling from the firing temperature of 1390°C, the honeycomb

structure, when the temperature became 320°C, was dropped into water (at 20°C) to apply a thermal shock.

In Example 38, Pt was supported onto the obtained honeycomb structure by PVD, specifically sputtering. The target was Pt, the sputtering gas was Ar, the reaction pressure was 1.3 Pa, the radio frequency was 13,56 MHz, the supplied power was 100W and the sputtering time period was 10 minutes. In Example 39, Pt was supported onto the obtained honeycomb structure using a super critical fluid. The super critical fluid used was CO₂ in which the honeycomb structure was immersed for 5 minutes. The amount of the supported Pt was 1.01 g/l in Example 38 and 1.55 g/l in Example 39.

Table 5 shows the characteristics of the honeycomb structures in Examples 37 to 39, evaluated in the same manner as described before. It is seen in Table 5 that oxygen vacancies were formed because of its oxygen content, number of oxygen and lattice constant in Example 37, and lattice defects were formed because of their oxygen content, number of oxygen and lattice constant in Examples 38 and 39.

When the honeycomb structures of Examples 37 to 39 were observed by TEM, a large number of fine cracks having a width of several nanometers or less were formed near the interfaces between the cordierite crystal phases and the amorphous phases of the honeycomb structure. Therefore, it is seen that when a cordierite-constituting element is replaced by an element having a different valence and a thermal shock is applied, both the fine pores by the oxygen vacancies or lattice defects as well as the fine pores by the fine cracks are formed.

In Examples 37 to 39, the thermal expansion coefficient was 0.59 to $0.75 \times 10^{-6}/^{\circ}\text{C}$, and the compressive strength in the direction of the flow channel was 10.1 to 10.6 MPa, both of which satisfy the requirements.

Table 5

Sample	Specific cordierite material	Firing atmosphere	Oxygen content (wt%)	Number of oxygen	Lattice constant b_0 (Å)	Temperature difference of thermal shock (°C)	Pt supported amount (g/L)	Thermal expansion coefficient ($\times 10^{-6}/^{\circ}\text{C}$)	Compressive strength (MPa)	Remarks
Example 37	Ga_2O_3	Air atmosphere	45.5	16.6	16.98	300	1.22	0.75	10.6	Si was replaced by Ga Thermal shock by re-heating
Example 38	GeO_2	Air atmosphere	48.5	17.7	17.01	300	1.01	0.61	10.4	Al was replaced by Ge Thermal shock during cooling step
Example 39	GeO_2	Air atmosphere	48.5	17.7	17.01	300	1.55	0.59	10.1	Al was replaced by Ge Thermal shock during cooling step

(Examples 40 to 42)

In Example 40, talc, kaolin, alumina and aluminum hydroxide were used as the Si, Al and Mg sources and were formulated into a near theoretical cordierite composition and the cordierite materials were further added with 0.05% by weight of NaCO₃ as a compound of an alkali metal element. The prepared cordierite materials were formed into a honeycomb shape, heated to remove a binder and then fired at 1390°C in an air atmosphere for 2 hours to produce a cordierite honeycomb structure, in the same manner as in Example 1. The obtained cordierite honeycomb structure was rapid cooled to room temperature by dropping into water (at 20°C) from 320°C in the course of cooling from the firing at 1390°C to apply a thermal shock.

Example 40 was repeated to produce a cordierite honeycomb structure, but the firing was carried out in a reduced pressure atmosphere having a pressure of 4000 MPa in Example 41 and in a reducing atmosphere of a hydrogen atmosphere in Example 42. The obtained cordierite honeycomb structures were cooled to room temperature (25°C) and then re-heated to 325°C and air was blown to apply a thermal shock.

The obtained honeycomb structures were measured for their thermal expansion coefficient and compressive strength. The results are shown in Table 6. In Examples 40 to 42, the thermal expansion coefficient was 0.42 to $0.58 \times 10^{-6}/^{\circ}\text{C}$, and the compressive strength in the direction of the flow channel was 10.8 to 12.1 MPa, both of which satisfy the requirements.

When the honeycomb structures of Examples 40 to 42 were observed by TEM, a large number of fine cracks having a width of several nanometers or less were formed near the interfaces between the cordierite crystal phases and the amorphous phases of the honeycomb structure.

The honeycomb structures of Examples 40 to 42 were immersed in a solution of 0.07 mol/l-chloroplatinic acid and 0.05 mol/l-rhodium chloride in ethanol while applying an ultrasonic wave for 10 minutes. The honeycomb structures were then blown with an air flow under 1×10^{-5} Pa (1 kgf/cm^2) to dry and then fired in air at 800°C for 2 hours. The thus Pt supported honeycomb structures were subjected to X-ray fluorescence spectroscopy to analyze the supported amounts of Pt and Rh. The honeycomb structure of Example 40 supported 1.2 g/l of Pt and 0.2 g/l of Rh, Example 41 supported 1.3 g/l of Pt and 0.3 g/l of Rh, and Example 42 supported 1.1 g/l of Pt and 0.2 g/l of Rh.

For comparison, Pt and Rh were similarly supported on the honeycomb structures of Examples 29 and 35, in which a thermal shock was applied but Na_2CO_3 was not added. The honeycomb structure of Example 29 supported 0.7 g/l of Pt and 0.2 g/l of Rh and Example 35 supported 0.5 g/l of Pt and 0.2 g/l of Rh, which are less than the amounts of the supported Pt and Rh in Examples 40 to 42. It is considered that in Examples 40 to 42, more fine cracks were formed by the thermal shock because the amorphous phase was increased by addition of Na_2CO_3 .

Table 6

Sample	Temperature difference of thermal shock (°C)	Catalyst supported amount (g/L)		Thermal expansion coefficient ($\times 10^{-6}/^{\circ}\text{C}$)	Compressive strength (MPa)	Remarks
		Pt	Rh			
Example 40	300	1.2	0.2	0.42	12.1	Na ₂ CO ₃ : 0.5 wt% addition Firing in air During cooling
						Na ₂ CO ₃ : 0.5 wt% addition Reduced pressure atmosphere Re-heating
Example 41	300	1.3	0.3	0.58	11.3	Na ₂ CO ₃ : 0.5 wt% addition Firing in air During cooling
						Na ₂ CO ₃ : 0.5 wt% addition Reduced pressure atmosphere Re-heating
Example 42	300	1.1	0.2	0.52	10.8	Na ₂ CO ₃ : 0.5 wt% addition Firing in reducing atmosphere Re-heating
						Firing in air
Example 43	-	1.9	0.2	0.38	10.6	Ultrasonic wave application

(Example 43)

As the cordierite materials, talc, kaolin, alumina and aluminium hydroxide were used and formulated into a near theoretical cordierite composition. In the same manner as in Example 1, the cordierite materials were added to a binder, a lubricant and a humidicant in appropriate amounts, formed into a honeycomb shape, and then fired in air at 1390°C for 2 hours, to obtain a honeycomb structure. A shock wave of a focused ultrasound with a frequency of 29 kHz and a power of 100W was applied to the obtained honeycomb structure. In the same manner as in Examples 40 to 42, catalyst components (Pt and Rh) were supported on the obtained cordierite honeycomb structure and the amounts of the supported Pt and Rh were measured. Also, the thermal expansion coefficient and the compressive strength in the direction of the flow channel were measured. The results are shown in Table 6.

As seen in Table 6, the amount of the supported Pt was 1.9 g/l and the amount of the supported Rh was 0.2 g/l.

The thermal expansion coefficient was $0.38 \times 10^{-6}/^{\circ}\text{C}$, and the compressive strength in the direction of the flow channel was 10.6 MPa, both of which satisfy the requirements.

By observing the honeycomb structure, it was confirmed that fine cracks having a width of a few nanometers were formed in a large number at interfaces between the cordierite crystal phases and the amorphous phases.

(Examples 44 to 46)

As the cordierite materials, talc, kaolin, alumina and aluminium hydroxide were used and formulated into a near theoretical cordierite composition. In the same manner as in Example 1, the cordierite materials were added to a binder, a lubricant and a humidicant in

appropriate amounts, formed into a honeycomb shape, and then fired in air at 1390°C for 2 hours, to obtain honeycomb structures.

The obtained honeycomb structure was immersed in
5 high temperature and high pressure water at 10 MPa and
300°C for 1 hour in Example 44. Similarly, the obtained
honeycomb structure was immersed in super critical fluid
CO₂ for 30 minutes in Example 45, and in an aqueous
solution of sodium hydroxide of 1 mol/l at 60°C for
10 5 hours in Example 46.

The amounts of the supported catalyst components (Pt and Rh), the thermal expansion coefficient and the compressive strength in the direction of the flow channel were measured. The results are shown in Table 7.

In Examples 44 to 46, the amount of the supported Pt
15 was 1.5 to 2.3 g/l, the amount of the supported Rh was
0.2 to 0.3 g/l, confirming that fine pores capable to support a catalyst component were formed. The thermal expansion coefficient was 0.35 to $0.39 \times 10^{-6}/^{\circ}\text{C}$, and the
20 compressive strength in the direction of the flow channel was 11.2 to 12.2 MPa, both of which satisfy the requirements.

Table 7

Sample	Treatment condition	Catalyst supported amount (g/L)		Thermal expansion coefficient ($\times 10^{-6}/^{\circ}\text{C}$)	Compressive strength (MPa)	Remarks
		Pt	Rh			
Example 44	High temperature and high pressure water (10 MPa, 300 °C)	1.5	0.3	0.39	11.9	Firing in air
Example 45	Super critical fluid (CO_2)	2.3	0.3	0.37	11.2	Firing in air
Example 46	Aqueous solution of sodium hydroxide (1 mol/L, 60 °C)	2.1	0.2	0.35	12.2	Firing in air

(Example 47 and 48)

As the cordierite materials, talc, kaolin, alumina and aluminium hydroxide were used and formulated into a near theoretical cordierite composition. In the same manner as in Example 1, the cordierite materials were added with a binder, a lubricant and a humidicant in appropriate amounts, formed into a honeycomb shape, and then fired in air at 1390°C for 2 hours, to obtain honeycomb structures.

In Example 47, the obtained honeycomb structure was dry etched with CF₄. The conditions of the dry etching was a CF₄ flow rate of 150 ml/min, a reaction chamber pressure of 13.3 Pa, an applied radio frequency of 13.56 MHz, a supplied power of 300W and an etching duration of 10 minutes. In Example 48, similarly, dry etching was carried out with Ar under the conditions of a reaction chamber pressure of 1.3 Pa, an applied radio frequency of 13.56 MHz, a supplied power of 100W and an etching duration of 10 minutes.

The amounts of the supported catalyst components (Pt and Rh), the thermal expansion coefficient and the compressive strength in the direction of the flow channel were measured. The results are shown in Table 8.

In Examples 47 to 48, the amount of the supported Pt was 1.1 to 1.3 g/l, and the amount of the supported Rh was 0.2 to 0.3 g/l, confirming that fine pores capable to support a catalyst component were formed. The thermal expansion coefficient was 0.45 to $0.46 \times 10^{-6}/^{\circ}\text{C}$, and the compressive strength in the direction of the flow channel was 11.7 to 12.7 MPa, both of which satisfy the requirements.

The results of the evaluation of Examples 47 and 48 are shown in Table 8.

Table 8

Sample	Treatment condition	Catalyst supported amount (g/L)		Thermal expansion coefficient ($\times 10^{-6}/^{\circ}\text{C}$)	Compressive strength (MPa)	Remarks
		Pt	Rh			
Example 47	Dry etching (CF_4)	1.1	0.3	0.46	11.7	Firing in air
Example 48	Sputter etching (Ar)	1.3	0.2	0.45	12.7	Firing in air

(Examples 49 to 51)

In Example 49, as the cordierite materials, talc, kaolin, alumina and aluminium hydroxide were used, but 5% by weight of the Si source was replaced by CeO₂, and 5 formulated into a near theoretical cordierite composition. In the same manner as in Example 1, the cordierite materials were added with a binder, a lubricant and a humidicant in appropriate amounts, formed 10 into a honeycomb shape, heated in air to 800°C to remove the binder and then fired in a reduced pressure atmosphere under 4000 MPa at 1390°C for 2 hours, to obtain a cordierite honeycomb structure.

In Example 50, a cordierite honeycomb structure was produced in the similar manner as in Example 49, but 5% by weight of the Al source, not the Si source, was 15 replaced by CeO₂ and the firing atmosphere was a reducing atmosphere of a hydrogen atmosphere.

In Example 51, a cordierite honeycomb structure was produced in the similar manner as in Example 49, but 5% by weight of the Mg source, not the Si source, was 20 replaced by CeO₂ and the firing atmosphere was an air atmosphere.

The obtained cordierite honeycomb structures were evaluated in their thermal expansion coefficient and the compressive strength in the direction of the flow 25 channel.

As seen in Table 9, the thermal expansion coefficient was 0.78 to $0.98 \times 10^{-6}/^{\circ}\text{C}$, and the compressive strength in the direction of the flow channel 30 was 10.8 to 12.1 MPa, both of which satisfy the requirements.

When catalyst components (Pt and Rh) were supported on the honeycomb structures of Examples 49 to 51, the amounts of the supported Pt and Rh were 1.5 to 2.3 g/l of 35 Pt and 0.2 to 0.3 g/l of Rh, confirming that fine pores capable of supporting a catalyst component were formed by

incorporating CeO₂ in the cordierite honeycomb structure. The amount of the supported catalyst components of Examples 49 to 51 is equivalent to that in the three way converter catalyst (1.5 g/l).

5

The results of the evaluation in Examples 49 to 51 are shown in Table 9.

Table 9

Sample	Replacing element	Oxygen storage capacity (O_2 mol/ CeO_2 mol)	Catalyst supported amount (g/L)	Compressive strength		Remarks
				Pt	Rh	
Example 49	GeO_2	4.2×10^{-2}	1.5	0.3	0.98	Si was replaced by Ce Firing in reduced pressure atmosphere
Example 50	GeO_2	3.1×10^{-2}	2.3	0.3	0.86	Al was replaced by Ce Firing in reducing atmosphere
Example 51	GeO_2	9.6×10^{-2}	2.1	0.2	0.78	Mg was replaced by Ce Firing in air

The oxygen storage capacity of the cordierite honeycomb structures of Examples 49 to 51 was evaluated. The evaluation of the oxygen storage capacity was made by TG-DTA (Rigaku: TAS-200). The produced Ce-substituted cordierite honeycomb structures were pulverized, and 5 20 mg of the pulverized powder was kept at 500°C, while an oxygen atmosphere of 50% of O₂, with the balance being N₂, and a reducing atmosphere of 2% of H₂, with the balance being N₂, were repeated twice. The amount of the 10 released oxygen was calculated from the weight difference between the oxygen atmosphere and the reducing atmosphere. This amount of the released oxygen was divided by the amount of the CeO₂ contained in the 15 cordierite honeycomb structure to obtain the amount of the released oxygen per 1 mol of CeO₂ contained in the cordierite honeycomb structure, that is, the oxygen storage capacity. The oxygen storage capacity is also 20 shown in Table 9.

For comparison, the oxygen storage capacity of a cordierite honeycomb structure without a Ce substitution 25 was also evaluated and that of a three way converter catalyst with CeO₂ supported thereon, in which the amount of the supported CeO₂ was made to be 75 g/l, 1.5 times the usual supported amount, was also evaluated. As a result, the cordierite honeycomb structure without a Ce substitution did not have an oxygen storage capacity and the three way converter catalyst with CeO₂ supported 30 thereon had an oxygen storage capacity of 1.5×10^{-2} O₂ mol/CeO₂ mol. In contrast, the oxygen storage capacities of Examples 49 to 51 were in a range of 3.1 to 9.6×10^{-2} O₂ mol/CeO₂ mol, which was larger than that of the three way catalyst.

The content of CeO₂ in Example 51, having the highest oxygen storage capacity, was only about 2% by 35 weight. The amount of the oxygen which can be stored by 1 liter of the cordierite honeycomb structure of

Example 51 and that by 1 liter of the three way catalyst (75 g/l of the supported CeO₂) were almost equal to each other. The experiment of Example 51, in which Mg was replaced by Ce, was repeated and the content of CeO₂ in the cordierite honeycomb structure of Example 51 was varied. The oxygen storage capacity was significantly reduced, to a level where it was difficult to detect, when the content of CeO₂ in the cordierite honeycomb structure was 0.01% by weight. Accordingly, at least 0.01% by weight of the content of CeO₂ is necessary to provide an oxygen storage capacity.

(Examples 52 and 53)

In Example 52, as the cordierite materials, talc, kaolin, alumina and aluminium hydroxide were used but 1.2% by weight of the Mg source was replaced by Pt, and they were formulated into a near-theoretical cordierite composition. In the same manner as in Example 1, the cordierite materials were added with a binder, a lubricant and a humidicant in appropriate amounts, formed into a honeycomb shape and then fired in an air atmosphere at 1390°C for 2 hours, to obtain a cordierite honeycomb structure.

The obtained cordierite honeycomb structure had a Pt content of 1.7 g/l. The cordierite honeycomb structure was evaluated in its thermal expansion coefficient and the compressive strength in the direction of the flow channel. As seen in Table 10, the thermal expansion coefficient was $0.85 \times 10^{-6}/^{\circ}\text{C}$, and the compressive strength in the direction of the flow channel was 10.9 MPa, both of which satisfy the requirements.

In Example 53, a cordierite honeycomb structure was produced in the similar manner as in Example 52, but 1.2% by weight of the Mg source was replaced with an oxide of Pt and 5% by weight of the Al source was replaced by CeO₂. After being cooled to room temperature, the

honeycomb structure was re-heated to 320°C and rapidly cooled by being dropped into water (20°C).

The obtained cordierite honeycomb structures had a Pt content of 1.6 g/l and an oxygen storage capacity of 5 $3.9 \times 10^{-2} O_2$ mol/CeO₂ mol. The cordierite honeycomb structure was evaluated in its thermal expansion coefficient and the compressive strength in the direction of the flow channel. As seen in Table 10, the thermal expansion coefficient was $0.96 \times 10^{-6}/^{\circ}C$, and the 10 compressive strength in the direction of the flow channel was 11.9 MPa, both of which satisfy the requirements.

The results of the evaluation of Examples 52 and 53 are shown in Table 10.

Table 10

Sample	Replacing element	Temperature difference of thermal shock (°C)	Oxygen storage capacity (O_2 mol/CeO ₂ mol)	Pt content (g/L)	Thermal expansion coefficient ($\times 10^{-6}/^{\circ}C$)	Compressive strength (MPa)	Remarks
Example 52	PtO	-	-	1.7	0.85	10.9	Mg was replaced by Pt Firing in air
Example 53	PtO GeO ₂	300	3.9×10^{-2}	1.6	0.96	11.9	Mg was replaced by Pt Al was replaced by Ce Firing in air Thermal shock by re-heating

(Cleaning test)

The performance of the thus produced cordierite honeycomb structures in cleaning hydrocarbon was evaluated.

5 A catalyst component was supported on each of the cordierite honeycomb structures produced in Examples 1, 11 to 18, 25 to 33 and 40 to 53, to form a catalyst-ceramic body. The catalyst-ceramic body was cut out into
10 a cylinder having a diameter of 15 mm and a length of 10 mm. The cylinder of the catalyst-ceramic body was used in the test of cleaning hydrocarbon at various predetermined temperatures in a model gas bench. The model gas comprised 500 ppm of C₃H₆, 5% of oxygen and the balance being nitrogen. The SV (space velocity) was
15 10,000/hour. The temperature at which 50% of C₃H₆ in the model gas was cleaned with a catalyst-ceramic body, was made as "50% cleaning temperature". The results are shown in Table 11.

Table 11

Sample	50% cleaning temperature (°C)	Catalyst supported amount (g/L)		Supporting method	Remarks
		Pt	Rh		
Comparative Example 8	489	-	-	-	No catalyst
Comparative Example 9	184	1.25	0.25	-	Three way converter catalyst
Example 1	188	1.9	-	Ethanol solution	
Example 11	263	7.7	-	Ethanol solution	
Example 12	233	2.3	-	Ethanol solution	
Example 13	255	8.8	-	Ethanol solution	
Example 14	221	2.5	-	Ethanol solution	
Example 15	238	2.4	-	Ethanol solution	
Example 16	263	2.0	-	Ethanol solution	
Example 17	238	4.4	-	Ethanol solution	
Example 18	187	2.4	-	Ethanol solution	
Example 25	223	1.2	-	CVD	
Example 26	233	1.0	-	PVD	
Example 27	212	1.5	-	Super critical fluid	
Example 28	204	3.7	-	Ethanol solution	
Example 29	206	4.1	-	Ethanol solution	
Example 30	190	2.6	-	Ethanol solution	
Example 31	194	4.1	-	Ethanol solution	
Example 32	206	3.4	-	Ethanol solution	
Example 33	195	3.8	-	Ethanol solution	
Example 40	221	1.2	0.2	Ethanol solution + ultrasonic wave	
Example 41	202	1.3	0.3	Ethanol solution + ultrasonic wave	
Example 42	206	1.1	0.2	Ethanol solution + ultrasonic wave	
Example 43	199	1.9	0.2	Ethanol solution + ultrasonic wave	
Example 44	207	1.5	0.3	Ethanol solution + ultrasonic wave	
Example 45	200	2.3	0.3	Ethanol solution + ultrasonic wave	
Example 46	230	2.1	0.2	Ethanol solution + ultrasonic wave	
Example 47	219	1.1	0.3	Ethanol solution + ultrasonic wave	
Example 48	211	1.3	0.2	Ethanol solution + ultrasonic wave	
Example 49	214	0.6	0.5	Ethanol solution + ultrasonic wave	
Example 50	226	1.8	1.1	Ethanol solution + ultrasonic wave	
Example 51	210	1.3	0.2	Ethanol solution + ultrasonic wave	
Example 52	253	1.7	-	-	
Example 53	247	1.6	-	-	

For comparison, the test device without the above catalyst-ceramic body was evaluated (Comparative Example 8), and a three way catalyst comprising a cordierite honeycomb structure with a coating of γ -alumina in an amount of 150 g/l on which 1/25 g/l of Pt and 0.25 g/l of Rh as well as 75 g/l of a co-catalyst CeO_2 were supported (Comparative Example 9) was also evaluated. The results are also shown in Table 11.

As seen in Table 11, the 50% cleaning temperature was high, 500°C, in the case of Comparative Example 8 in which no catalyst-ceramic body was used, and was lowered to 184°C in the case of the three way catalyst in Comparative Example 9. In contrast, in the cases of Examples of the present invention in which the amount of the supported catalyst component was as much as that of the three way catalyst, the 50% cleaning temperature was equivalent to or slightly higher than that of the three way catalyst, and was about a half of that of the case of non-catalyst-ceramic body. Thus, a high cleaning capability was attained in the Examples.

The relationship between the cleaning capability and the state of the supported catalyst component was examined as below:

Cordierite materials, in which 10% by weight of the Si source was replaced by an oxide of an element having a valence different from Si, Ga_2O_3 , was prepared, formed into a honeycomb shape, and fired in air at 1390°C for 2 hours. To the obtained honeycomb structure, catalyst components were supported by immersing in an ethanol solution of 0.07 mol/l of chloroplatinic acid and 0.05 mol/l of rhodium chloride, while ultrasound was being applied, for 10 minutes, drying and firing in air at 800°C for 2 hours, to form a catalyst-ceramic body (Sample 1). A similar sample prepared without applying ultrasound was made as Sample 2. Sample 2 was heat

treated at 1000°C for 50 hours and the thus heat deteriorated sample was made as Sample 3.

The Samples 1 to 3 were observed by TEM to examine the state of the supported catalyst metal particles. The 5 TEM magnification was $\times 200,000$. The spacing between the catalyst metal particles was measured for 30 particles in one field of view and the measurement was conducted for 5 fields of view to obtain an average spacing between the catalyst metal particles. The results are shown in 10 Table 12.

The 50% cleaning temperature of C_3H_6 was also measured for Samples 1 to 3. They were 254°C, 336°C and 460°C.

Similarly, the same ceramic support as in Samples 1 15 to 3 was used and immersed in an aqueous solution of 0.005 mol/l of chloroplatinic acid and 0.003 mol/l of rhodium chloride, dried and fired in air at 800°C for 2 hours, to form a catalyst-ceramic body (Sample 4). Also, the same ceramic support was immersed in an aqueous 20 solution of 0.0025 mol/l of chloroplatinic acid and 0.0015 mol/l of rhodium chloride, dried and fired in air at 800°C for 2 hours, to form a catalyst-ceramic body (Sample 5). No ultrasound was applied for Samples 4 and 5 during the supporting of the catalyst component.

25 The Samples 4 and 5 were observed by TEM at the magnitude of $\times 50,000$. The spacing between the catalyst metal particles was measured in the same manner as for Samples 1 to 3 and the results are also shown in Table 12.

30

Table 12

Sample No.	1	2	3	4	5
50% cleaning temperature (°C)	254	336	460	472	484
Average spacing between catalyst metal particles (nm)	23	74	108	850	1009

In Table 12, Sample 3, which was deteriorated by heating, had a 50% cleaning temperature of C₃H₆ of 460°C and an average spacing between the catalyst metal particles of 108 nm, beyond 100 nm. As seen in Table 11, 5 the 50% cleaning temperature of C₃H₆ by a non-catalyst-ceramic body was 484°C and, therefore, Sample 3 is considered to have a cleaning capability.

Sample 2 had a 50% cleaning temperature of C₃H₆ of 336°C and an average spacing between the catalyst metal 10 particles of 74 nm, while Sample 1 had a 50% cleaning temperature of C₃H₆ of 254°C and an average spacing between the catalyst metal particles of 23 nm. It is seen that as in the average spacing between the catalyst metal particles decreased, the 50% cleaning temperature 15 lowered.

Sample 4, having an average spacing between the catalyst metal particles of less than 1000 nm, had a 50% cleaning temperature of C₃H₆ of 472°C, lower than 484°C of the non-catalyst-ceramic body, indicating that 20 Sample 4 had a cleaning capability.

However, Sample 5, having an average spacing between the catalyst metal particles of more than 1000 nm, had a 50% cleaning temperature of 484°C, equal to that of the non-catalyst-ceramic body, indicating that Sample 5 did 25 not have a cleaning capability.

Therefore, it can be seen that the average spacing between the catalyst metal particles should be not more than 1000 nm, preferably not more than 100 nm to have a cleaning capability. With a smaller average spacing 30 between the catalyst metal particles, the cleaning capability is improved. When using ultrasound during the supporting step, the catalyst component can be more easily infiltrated in the fine pores of the ceramic support and the average spacing between the catalyst 35 metal particles can be decreased.

CLAIMS

1. A ceramic support which comprises a crystal lattice having at least one of oxygen vacancy and lattice defects therein, by which said ceramic support can support a catalyst component.
5
2. The ceramic support according to claim 1, wherein said ceramic support has a shape of at least one of a honeycomb, a pellet and a powder.
3. A ceramic support which has cracks by which
10 said ceramic support can support a catalyst component.
4. The ceramic support according to claim 4, wherein said cracks are present in at least one of the crystal and the amorphous phases.
5. The ceramic support according to claim 3,
15 wherein said cracks have a width in a range of 100 nm or less.
6. The ceramic support according to claim 3,
20 wherein said ceramic support has a shape of at least one of a honeycomb, a pellet and a powder.
7. A ceramic support capable of supporting a catalyst, comprising a ceramic body having fine pores with a diameter or width of up to 1000 times the ion diameter of a catalyst component to be supported on the surface of said ceramic body, the number of said fine pores being not less than 1×10^{11} pores per liter.
25
8. The ceramic support according to claim 7,
30 wherein said ceramic body has five pores with a diameter or width of 1 to 1000 times the ion diameter of a catalyst component to be supported on the surface of said ceramic body.
9. The ceramic support according to claim 7,
wherein said number of said fine pores is not less than 1×10^{16} pores per liter.
10. The ceramic support according to claim 7,
35 wherein said fine pores are composed of at least ones of oxygen vacancies and lattice defects in the crystal

lattice of said ceramic, fine cracks on the surface of said ceramic body, and vacancies of an element or elements constituting the ceramic.

5 11. The ceramic support according to claim 7, wherein said ceramic mainly comprises cordierite.

12. The ceramic support according to claim 7, wherein said ceramic support is in the form of a honeycomb.

10 13. The ceramic support according to claim 7, wherein said ceramic support has a compressive strength in the direction of the flow channel of not less than 1 MPa and a thermal expansion coefficient in the direction of the flow channel of not higher than $1.0 \times 10^{-6}/^{\circ}\text{C}$.

15 14. The ceramic support according to claim 13, wherein said ceramic support has a compressive strength in the direction of the flow channel of not less than 10 MPa.

20 15. The ceramic support according to claim 7, wherein said ceramic support has the fine pores in a number of not less than 1×10^{17} pores per liter.

25 16. The ceramic support according to claim 7, wherein said diameter or width of said fine pores is in a range of 0.1 to 100 nm.

17. The ceramic support according to claim 7, wherein said fine pores have a depth of not smaller than a half of the ion diameter of a catalyst component to be supported on the surface of said ceramic body.

30 18. The ceramic support according to claim 11, wherein said ceramic has a content of oxygen in an amount of less than 47% by weight or more than 48% by weight.

35 19. The ceramic support according to claim 11, wherein said ceramic comprises cordierite crystals and the lattice constant of b_0 axis of the cordierite crystals is larger than 16.99 or smaller than 16.99.

20. The ceramic support according to claim 11,

wherein said ceramic includes cordierite crystals having one or more of at least one of oxygen vacancy and lattice defect in the unit lattice of cordierite crystal in an amount of not less than $4 \times 10^{-8}\%$ of said ceramic.

5 21. The ceramic support according to claim 20, wherein said ceramic includes said cordierite crystals having one or more of at least one of oxygen vacancy and lattice defect in the unit lattice of cordierite crystal in an amount of not less than $4 \times 10^{-5}\%$ of said ceramic.

10 22. The ceramic support according to claim 7, wherein said ceramic comprises cordierite crystals which comprise not less than 4×10^{-8} of at least one of oxygen vacancy and lattice defect per unit lattice of cordierite crystal.

15 23. The ceramic support according to claim 22, wherein said ceramic comprises cordierite crystals which comprise not less than 4×10^{-7} of at least one of oxygen vacancy and lattice defect per unit lattice of cordierite crystal.

20 24. A ceramic support capable of supporting a catalyst component, comprising a honeycomb structure mainly comprised of a cordierite composition and having oxygen vacancies in cordierite crystals, said honeycomb structure having a content of oxygen in an amount of less than 47% by weight.

25 25. The ceramic support according to claim 24, wherein said oxygen vacancies are formed by firing in a reduced pressure atmosphere, a reducing atmosphere, or a low oxygen concentration atmosphere.

30 26. The ceramic support according to claim 25, wherein said firing atmosphere is a reduced pressure atmosphere at a pressure of not higher than 4000 Pa, a reducing atmosphere, or a low oxygen concentration atmosphere with a oxygen concentration in a range of not less than 0% and up to less than 3%.

35 27. The ceramic support according to claim 24,

wherein said oxygen vacancies are formed by replacing at least some of Si, Al and Mg elements constituting the cordierite by an element having a valency smaller than that of the replaced element.

5 28. The ceramic support according to claim 24, wherein said cordierite crystals have a lattice constant of b_0 axis of smaller than 16.99.

10 29. A ceramic support capable of supporting a catalyst component, comprising a honeycomb structure mainly comprised of a cordierite composition and having lattice defects in cordierite crystals, said honeycomb structure having a content of oxygen in an amount of more than 48% by weight.

15 30. The ceramic support according to claim 29, wherein said lattice defects are formed by replacing at least some of Si, Al and Mg elements constituting the cordierite by an element having a valency greater than that of the replaced element.

20 31. The ceramic support according to claim 29, wherein said cordierite crystals have a lattice constant of the b_0 axis of larger than 16.99 or smaller than 16.99.

25 32. A ceramic support capable of supporting a catalyst component, comprising a honeycomb structure mainly comprised of a cordierite composition and having a multiple number of fine cracks in at least one of the amorphous and the crystal phases thereof.

30 33. The ceramic support according to claim 32, wherein said ceramic support contains an alkali metal and an alkaline earth metal in a total amount of not less than 0.05% by weight.

35 34. The ceramic support according to claim 32, wherein said fine cracks have a width of not more than 100 nm and said fine cracks are formed by thermal shock or shock waves.

35. A ceramic support capable of supporting a catalyst component, comprising a honeycomb structure

mainly comprised of a cordierite composition and having at least ones of oxygen vacancies and lattice defects as well as having a multiple number of fine cracks in at least one of the amorphous and the crystal phases thereof.

5

36. The ceramic support according to claim 32 or 35, wherein said honeycomb structure includes said oxygen vacancies, lattice defects and fine cracks in a total number amount of at least 1×10^{16} per liter.

10

37. A ceramic support capable of supporting a catalyst component, comprising a honeycomb structure mainly comprised of a cordierite composition, said honeycomb structure having an oxygen content of less than 47% by weight or more than 48% by weight and a multiple number of fine cracks in at least one of the amorphous and the crystal phases thereof.

15

38. A ceramic support capable of supporting a catalyst component, comprising a honeycomb structure mainly comprised of a cordierite composition, a lattice constant of b_0 axis of the cordierite crystals of said honeycomb structure being larger than 16.99 or smaller than 16.99, said honeycomb structure having a multiple number of fine cracks in at least one of the amorphous and the crystal phases thereof.

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39. A process for producing a ceramic support capable of supporting a catalyst component, comprising the step of:

30

preparing cordierite materials comprising an Si source, an Al source and a Mg source as well as a binder,

forming said cordierite materials into a honeycomb shape,

heating said honeycomb shape to remove said binder and then

35

firing said honeycomb shape in a reduced pressure or reducing atmosphere to form a honeycomb structure mainly comprised of a cordierite composition.

40. The process according to claim 39, wherein said firing atmosphere is a reduced pressure atmosphere at a pressure of not higher than 4000 Pa, or a reducing atmosphere.

5 41. A process for producing a ceramic support capable of supporting a catalyst component, comprising the step of:

10 preparing cordierite materials comprising an Si source, an Al source and a Mg source as well as a binder, at least some of said Si, Al and Mg sources being used from compounds which do not comprise oxygen,

15 forming said cordierite materials into a honeycomb shape,

15 heating said honeycomb shape to remove said binder and then

20 firing said honeycomb shape in a low oxygen concentration atmosphere with an oxygen concentration in a range of not less than 0% to less than 3% to form a honeycomb structure mainly comprised of a cordierite composition.

25 42. A process for producing a ceramic support capable of supporting a catalyst component, comprising the step of:

25 preparing cordierite materials comprising an Si source, an Al source and a Mg source as well as a binder, some of said Si, Al and Mg sources being replaced by compounds comprising an element having a valency smaller than that of the replaced Si, Al or Mg,

30 forming said cordierite materials into a honeycomb shape,

30 heating said honeycomb shape to remove said binder and then

35 firing said honeycomb shape in a reduced pressure atmosphere, a reducing atmosphere, an oxygen-containing atmosphere or an oxygen-free atmosphere to form a honeycomb structure mainly comprised of a cordierite composition.

43. The process according to claim 42, wherein said firing atmosphere is a reduced pressure atmosphere at a pressure of not higher than 4000 Pa, a reducing atmosphere, an oxygen-containing atmosphere or an oxygen-free atmosphere.

5 44. A process for producing a ceramic support capable of supporting a catalyst component, comprising the step of:

10 preparing cordierite materials comprising an Si source, an Al source and a Mg source as well as a binder, some of said Si, Al and Mg sources being replaced by compounds comprising an element having a valency larger than that of the replaced Si, Al or Mg,

15 forming said cordierite materials into a honeycomb shape, and

 firing said honeycomb shape in air to form a honeycomb structure mainly comprised of a cordierite composition.

20 45. A process for producing a ceramic support capable of supporting a catalyst component, comprising the step of:

25 preparing cordierite materials comprising an Si source, an Al source and a Mg source as well as a binder,

 forming said cordierite materials into a honeycomb shape,

 firing said honeycomb shape to form a honeycomb structure mainly comprised of a cordierite composition, and

30 heating said honeycomb structure to a predetermined temperature followed by rapid cooling said honeycomb shape from said predetermined temperature.

35 46. A process for producing a ceramic support capable of supporting a catalyst component, comprising the step of:

 preparing cordierite materials comprising an Si source, an Al source and a Mg source as well as a

binder,

forming said cordierite materials into a honeycomb shape,

firing said honeycomb shape, and

5 rapidly cooling said fired honeycomb shape from a predetermined temperature during cooling from the temperature used in said firing to form a honeycomb structure mainly comprised of a cordierite composition.

10 47. The process according to claim 39, 41, 42, 44, 45 or 46, wherein said obtained fired honeycomb structure is further heated to a predetermined temperature and then rapidly cooled from said predetermined temperature.

15 48. The process according to claim 39, 41, 42, 44, 45 or 46, wherein said obtained fired honeycomb structure is further rapidly cooled from a predetermined temperature during cooling from the temperature used in said firing.

20 49. The process according to claim 45 or 46, wherein a temperature difference between said predetermined temperature and a temperature after said rapid cooling is not more than 900°C.

50. The process according to claim 39, 41, 42, 44, 45 or 46, wherein said obtained fired honeycomb structure is further subjected to a shock wave.

25 51. The process according to claim 50, wherein said shock wave is provided by ultrasound or vibration.

52. The process according to claim 39, 41, 42, 44, 45 or 46, wherein a compound of an alkali metal element or alkaline earth metal element is added to said

30 cordierite materials.

53. A process for producing a ceramic support capable of supporting a catalyst component, comprising the step of:

35 preparing cordierite materials comprising an Si source, an Al source and a Mg source as well as a binder,

forming said cordierite materials into a

honeycomb shape,

firing said honeycomb shape to form a honeycomb structure mainly comprised of a cordierite composition, and

5 immersing said honeycomb structure in a high temperature and high pressure water, a super critical fluid or an alkali solution.

10 54. A process for producing a ceramic support capable of supporting a catalyst component, comprising the step of:

15 preparing cordierite materials comprising an Si source, an Al source and a Mg source as well as a binder,

20 forming said cordierite materials into a honeycomb shape,

25 firing said honeycomb shape to form a honeycomb structure mainly comprised of a cordierite composition, and

30 dry etching or sputter etching said honeycomb structure.

55. A ceramic support capable of supporting a catalyst component, comprising a honeycomb structure and comprising a substance having an oxygen storage capacity.

56. The ceramic support according to claim 55, wherein said ceramic support comprises CeO₂ as said substance having an oxygen storage capacity in an amount of not less than 0.01% by weight.

57. The ceramic support according to claim 55, wherein said honeycomb structure is mainly comprised of a cordierite composition.

58. A ceramic support capable of supporting a catalyst component, comprising a honeycomb structure mainly comprised of a cordierite composition, in which some of at least one of Si, Al and Mg elements constituting cordierite is replaced by Ce.

59. The ceramic support according to claim 7, 32, 35, 37 or 38, wherein a co-catalyst having an oxygen

storage capacity is supported by said ceramic support without a coating layer being formed on the surface of said ceramic support, to provide an oxygen storage capacity.

5 60. The ceramic support according to claim 7, 32, 35, 37 or 38, wherein a precursor to a co-catalyst having an oxygen storage capacity is provided to said ceramic support without a coating layer being formed on the surface of said ceramic support, and said ceramic support with said precursor is then heated to provide an oxygen storage capacity to said ceramic support.

10 61. The ceramic support according to claim 55 or 58, wherein said cordierite includes at least ones of oxygen vacancies and lattice defects in the cordierite crystal lattice.

15 62. The ceramic support according to claim 55 or 58, wherein said ceramic support has a multiple number of fine cracks in at least one of the amorphous and the crystal phases thereof.

20 63. The ceramic support according to claim 55 or 58, wherein said fine cracks have a width of not more than 100 nm.

25 64. A process for producing a ceramic support capable of supporting a catalyst component, comprising the step of:

30 preparing cordierite materials comprising an Si source, an Al source and a Mg source as well as a binder, some of said Si, Al and Mg sources being replaced by a Ce-containing compound,

35 forming said cordierite materials into a honeycomb shape,

 heating said honeycomb shape to remove said binder, and

 firing said honeycomb shape in a reduced pressure atmosphere at a pressure of not higher than 4000 Pa, a reducing atmosphere, an oxygen-containing atmosphere or an oxygen-free atmosphere to form a

honeycomb structure mainly comprised of a cordierite composition.

5 65. The process according to claim 64, wherein said obtained fired honeycomb structure is further heated to a predetermined temperature and then rapidly cooled from said predetermined temperature.

10 66. The process according to claim 64, wherein said obtained fired honeycomb structure is further rapidly cooled from a predetermined temperature during cooling after said firing.

15 67. The process according to claim 65, wherein a temperature difference between said predetermined temperature and a temperature after said rapid cooling is not more than 900°C.

20 68. The process according to claim 64, wherein said obtained fired honeycomb structure is further subjected to a shock wave.

25 69. The process according to claim 68, wherein said shock wave is provided by ultrasound or vibration.

20 70. The ceramic support according to claim 39, wherein a co-catalyst having an oxygen storage capacity is supported by said ceramic support without a coating layer being formed on the surface of said ceramic support, to provide an oxygen storage capacity.

25 71. The ceramic support according to claim 39, wherein a precursor to a co-catalyst having an oxygen storage capacity is provided to said ceramic support without a coating layer being formed on the surface of said ceramic support, and said ceramic support with said precursor then being heated to provide an oxygen storage capacity to said ceramic support.

30 72. A catalyst-ceramic body, comprising a catalyst component supported by said ceramic support as set forth in any one of claims 1 to 38 and 55 to 63.

35 73. A catalyst-ceramic body comprising a catalyst component directly supported on a ceramic support without a coating layer on the surface thereof.

74. The catalyst-ceramic body according to
claim 73, wherein said catalyst-ceramic body comprises a
catalyst component in an amount as the metal element of
not less than 0.01% by weight, an average distance
5 between particles of said catalyst component on the
surface of said ceramic support being in a range of 0.1
to 1000 nm.

75. The catalyst-ceramic body according to claim 73
or 74, wherein said average distance between particles of
10 said crystal component is in a range of 0.1 to 100 nm.

76. The catalyst-ceramic body according to
claim 73, wherein said catalyst component is at least one
of metals having a catalyst activity and metal oxides
having a catalyst activity.

77. The catalyst-ceramic body according to
claim 76, wherein said metals having a catalyst activity
are noble metals and said metal oxides having a catalyst
activity are oxides containing at least one metal
selected from the group of V, Nb, Ta, Cr, Mo, W, Mn, Fe,
20 Co, Ni, Cu, Zn, Ga, Sn and Pb.

78. The catalyst-ceramic body according to
claim 72, wherein said ceramic support has a multiple
number of fine pores with a diameter or width of 0.1 to
100 nm on the surface thereof.

79. The catalyst-ceramic body according to any one
of claims 73 to 78, wherein said ceramic support is the
ceramic support as set forth in any one of claims 7 to 38
and 55 to 63.

80. A catalyst-ceramic body comprising a ceramic
30 support comprising a honeycomb structure mainly comprised
of a cordierite composition, some of at least one of Si,
Al and Mg elements constituting the cordierite being
replaced by a metal having a catalyst activity.

81. The catalyst-ceramic body according to
35 claim 80, wherein said catalyst-ceramic body contains a
metal having a catalyst activity in an amount of not less
than 0.01% by weight thereof.

82. The catalyst-ceramic body according to
claim 80, wherein said catalyst-ceramic body contains a
metal having a catalyst activity in an amount of not less
than 0.01% by weight thereof and CeO₂ in an amount of not
less than 0.01% by weight thereof.

5 83. The catalyst-ceramic body according to
claim 80, wherein said metal having a catalyst activity
is at least one of noble metals, V, Nb, Ta, Cr, Mo, W,
Mn, Fe, Co, Ni, Cu, Zn, Ga, Sn and Pb.

10 84. The catalyst-ceramic body according to
claim 80, wherein said honeycomb structure has at least
ones of oxygen vacancies and lattice defects in the
cordierite crystal lattice, on which a catalyst component
is supported.

15 85. The catalyst-ceramic body according to
claim 80, wherein said honeycomb structure has a multiple
number of fine cracks in at least one of the amorphous
and the crystal phases thereof, on which a catalyst
component is supported.

20 86. The catalyst-ceramic body according to
claim 85, wherein said fine cracks have widths of not
more than 100 nm.

25 87. A process for producing a catalyst-ceramic
body, comprising the step of:

 preparing cordierite materials comprising
an Si source, an Al source and a Mg source as well as a
binder, some of said Si, Al and Mg sources being replaced
by a noble metal-containing compound,

30 forming said cordierite materials into a
honeycomb shape,

 heating said honeycomb shape to remove
said binder, and

35 firing said honeycomb shape in a reduced
pressure atmosphere at a pressure of not higher than
4000 Pa, a reducing atmosphere, an oxygen-containing
atmosphere or an oxygen-free atmosphere to form a
catalyst-ceramic body comprising a ceramic support of a

honeycomb structure mainly comprised of a cordierite composition.

88. A process for producing a catalyst-ceramic body, comprising the step of:

5 preparing cordierite materials comprising an Si source, an Al source and a Mg source as well as a binder, some of said Si, Al and Mg sources being replaced by a noble metal-containing compound and a Ce-containing compound,

10 forming said cordierite materials into a honeycomb shape,

 heating said honeycomb shape to remove said binder, and

15 15 firing said honeycomb shape in a reduced pressure atmosphere at a pressure of not higher than 4000 Pa, a reducing atmosphere, an oxygen-containing atmosphere or an oxygen-free atmosphere to form a catalyst-ceramic body comprising a ceramic support of a honeycomb structure mainly comprised of a cordierite composition.

20 20 89. The process according to claim 87, wherein said obtained fired honeycomb structure is further heated to a predetermined temperature and then rapidly cooled from said predetermined temperature.

25 25 90. The process according to claim 87, wherein said obtained fired honeycomb structure is further rapidly cooled from a predetermined temperature during cooling from the temperature used in said firing.

30 30 91. The process according to claim 89, wherein a temperature difference between said predetermined temperature and the temperature after said rapid cooling is not more than 900°C.

35 92. The process according to claim 87, wherein said obtained fired honeycomb structure is further subjected to a shock wave.

93. The process according to claim 92, wherein said shock wave is provided by ultrasound or vibration.

94. A process for producing the catalyst-ceramic body as set forth in claim 72, comprising depositing a catalyst component and/or a precursor of a catalyst component on said ceramic support by a CVD or PVD method.

5 95. A process for producing the catalyst-ceramic body as set forth in claim 72, comprising depositing a catalyst component and/or a precursor of a catalyst component on said ceramic support by means of a super critical fluid.

10 96. A process for producing the catalyst-ceramic body as set forth in claim 72, comprising depositing a catalyst component and/or a precursor of a catalyst component on said ceramic support by means of a solvent having a surface tension smaller than water.

15 97. A process for producing the catalyst-ceramic body as set forth in claim 72, comprising depositing a catalyst component and/or a precursor of a catalyst component on said ceramic support by means of a solvent having a surface tension smaller than water while applying vibration or performing vacuum defoaming.

20 98. A process for producing the catalyst-ceramic body as set forth in claim 72, comprising depositing a precursor of a catalyst component on said ceramic support followed by a heat treatment.

25 99. A process for producing the catalyst-ceramic body as set forth in claim 72, comprising depositing a catalyst component a plurality of times using the same or different compositions.

30 100. The ceramic support according to any one of claims 11 to 38 and 57 to 63, wherein said cordierite has a theoretical composition expressed by $2\text{MgO}\cdot 2\text{Al}_2\text{O}_3\cdot 5\text{SiO}_2$.

35 101. The process for producing a ceramic support according to any one of claims 39 to 54 and 64 to 71, wherein said cordierite has a theoretical composition expressed by $2\text{MgO}\cdot 2\text{Al}_2\text{O}_3\cdot 5\text{SiO}_2$.

102. The catalyst-ceramic body according to any one of claims 80 to 86, wherein said cordierite has a

theoretical composition expressed by $2\text{MgO} \cdot 2\text{Al}_2\text{O}_3 \cdot 5\text{SiO}_2$.

103. The process for producing a catalyst-ceramic body according to any one of claims 89 to 99, wherein said cordierite has a theoretical composition expressed
5 by $2\text{MgO} \cdot 2\text{Al}_2\text{O}_3 \cdot 5\text{SiO}_2$.

A CERAMIC SUPPORT CAPABLE OF SUPPORTING A CATALYST,
A CATALYST-CERAMIC BODY AND PROCESSES FOR PRODUCING SAME

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ABSTRACT OF THE INVENTION

10 A ceramic support capable of supporting a catalyst, comprising a ceramic body having fine pores with a diameter or width of up to 1000 times the ion diameter of a catalyst component to be supported on the surface of said ceramic body, the number of said fine pores being not less than 1×10^{11} pores per liter, is produced by
15 introducing oxygen vacancies or lattice defects in the cordierite crystal lattice or by applying a thermal shock to form fine cracks.

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Fig. 1

